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IN TWO SECTIONS

*Section One*



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## AVIATION

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## Diversifying Terminal Facilities

NEW YORK CITY has two railroad stations within the boundaries of Manhattan, and several others barely across the Hudson River. There are few travelers resident in the eastern part of this country whose experience has not included an occasional hurried dash by train to the nation's capital to make a connection when arrived at the Grand Central has provided by but twenty minutes or so the necessary hour of departure from the Pennsylvania, or vice versa. There are few dwellers in the neighborhood of Albany who have not given way to occasional exasperation at the impossibility of securing a through car to the south or of checking luggage to any point beyond New York without arranging,—and paying,—for a transfer through the city.

Other metropolitan centers, finding themselves in glass houses, have no privilege of casting stones. Detroit, Chicago, Baltimore and Montreal have the same multiplicity of rail terminals. London with close to a dozen major stations, presents a complexity even worse than any American city. With so distressing an object lesson before them, the operators of air transport have not been expected as a matter of course to pick a common terminal in each city, where passengers, baggage, and mail could be transhipped without waste of time and without inconvenience. The facts do not fit the expectation.

Los Angeles, with eight air lines under seven independent managements, already has them flying out of five different fields. Other busy western centers show a tendency in the same direction. Some cities not so busy still split up their terminal facilities to a ridiculous degree.

There is one consequence of very modern use through which two air lines pass, each one running a single plane in each direction daily. There is at least a theoretical opportunity of transfer from one to the other. If it remains theoretical it is partly because the companies

concerned have selected to use separate fields a couple of miles apart.

We cannot too grossly apply the financial estimate that has provided the funds for such a multitude of transport fields of such wide dimensions and splendid quality, and we heartily appreciate the thought that has gone into their planning and, in many instances, the exceptional artistic value of the appropriate buildings. Nevertheless we view this extreme separation with some little alarm and a great deal of regret, for at least two reasons.

In the first place, it is a drain upon air transport's financial resources. The transport companies themselves obviously ought to be the best judges of what they can afford, and in some cases the field overhead is clarified by other operations than those of straight transportation, but the building of magnificent passenger terminals, any one of which would be capable of handling several times the aggregate traffic in sight for some time to come, is inherently wasteful. There has to be a corresponding increase of income somewhere along the line. The total load that the public has to be persuaded to carry in order that air transport may be a financial success is proportionately increased.

Of more direct importance to the traveler is his own convenience. With the further development of the transport map, and the filling out of schedules with more frequent departures, the competition between air lines will appear more conspicuous compared with their cooperation. Inactive tickets will be a commonplace. Every company will benefit by simplifying the process of transfer and minimizing the loss of time involved in accomplishing it.

Finally there is a problem of control, which will be a super one. While collision hazards might be reduced by doing all flying to and from a single point if the plans drew perfectly blind, they will be much reduced by being kept under a wholly unified radio control.

When conditions are bad, all the aircraft converging on a city should be immediately under the command of a single official, with all the communication at his fingertips. There is no other way of safety. If radio-voice circuits have to be maintained on several parallel courses between the same cities leading to but only a few miles apart, confusion will become understandable.

We do not deny that there are circumstances in which the local circumstances are so peculiar that the regular use of several circuits, as opposed to the more emergency use of secondary fields when the regular one is covered with fog, may be desirable, but we believe that such cases are few. We believe that communication is definitely in the general interest in matters such as that of today. We speak for the passengers of the future and for air transport progress in expressing the hope that during the next few years, at least, every possibility of agreeing will be exhausted before more individual companies strike out on independent lines and without regard to their neighbors.

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### Seaplane Research

AMERICA is awake to the seaplane at last. After 15 years in which the voices of a few prophets crying in the wilderness seemed unnoticed—years when even the Navy seemed content to allow the art of flying boat operation to perish along with the last F5L, and one-field new design and construction to the annual production of one or two new PNs at the Naval Aircraft Factory—flying boat factories suddenly spring up on every hand. Not able to wait for the evolution of American designs in addition to the very few that have already appeared themselves, promoters and financiers have coaxed Europe for promising types. Already we have driven on the products of Germany, France, Italy, and Great Britain, all of them either already being built in the United States or planned for construction here in the near future.

All that is a commonplace. It may seem too old a story to be worth repeating editorially, but it is easy to take it too much for granted. Assuming that we have a flying-boat industry, and that it is here to stay, what next? How shall we guide its growth? We cannot resign ourselves to inertia permanently; quiet followers is the fate of the Dornier and Blohm, of the Schmeidler and Short. We cannot rely permanently upon the constraining commercial design to drive parallelism with the more or less standard hull forms adopted by the Navy, excellent as it is in record and cost as is the credit that it sheds upon Curtiss, Richardson and his co-workers in its evolution. We should restrict our progress in a very measure if we devoted to the purely commercial builder, having no direct relation with the naval service

and seeking only the opportunity of developing its own originality and displaying its own talents under the most favorable conditions.

Certainly it is impossible, especially when the focusing of so much attention upon the giant seaplane, to depend on trial and error, or to hazard the whole fate of the sea interest in building a new type upon the efficiency and shortcomings of a novel form of hull based purely upon speculation.

To the fundamental question, as to many others of the same type, there is only one answer. Research holds the key—and research upon seaplanes cannot be towing lawn, or water channel. By it we find the regular parallel for the wild trail. By it we see we are in position to determine quality and cheaply and safely and accurately the peculiarities of behavior of a hull upon the water, heeling, or tilting off. We can evolve at least empirical, if not rational, laws of designing for specified characteristics of action on the surface.

For nearly twenty years seaplane hull studies have gone on in Washington, handicapped by the limited space of momentary possible in the Navy Yard basin and by the inadequacy of refinements designed for a much heavier class of work handicapped by pressure upon the capacity of equipment and staff and by the necessity of competing for a place on the schedule with studies of hydrotype and streamlines. Now, for the first time in this country, there is to be a laboratory having seaplane research in its first aim.

The aircraft industry does not yet seem to appreciate what cause for rejoicing there is in this fact. Those who see a great future for the seaplane and recognize the need of constant forward progress in the design of the type should find an eagerly important attention on the National Advisory Committee for Aeronautics. They still will have some time to wait, but progress is being made. The day when the Commission opens the new high-speed water channel at Langley Field, making it available for research and for the testing of the industry's own products, will exceed in potential importance—no say it—sharply and conservatively—any other in American seaplane history in the last ten years.

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### Seaplane Not Pilots

IT IS AN AXIOM of selling that nothing can equal face-to-face personal contact. Barring certain types of mail order business, almost all selling is done by word of mouth between two individuals, and much as direct mail and other forms of advertising may help, it is the personal appeal that wins success on the dotted line.

Aircraft selling is no exception to this rule and the need for more and better aircraft salesmen has been and is being felt. There has been much contention as to

whether or not aircraft salesmen should first be pilots, the chief argument in those of pilot salesmen being that only a pilot can speak the language of the air. On the other hand it must be admitted that most pilots—salesmen are pilots first and salesmen second. And in those days of increasing sales competition a salesman should be able to concentrate all his time on selling only.

It must be admitted that aviation as an industry does not advance greatly until big business steps in, and it seems now that sales organizations might profit by an infusion of fresh blood. After all, in selling aircraft and their services it is more important to talk to business men in their own language, that of dollars and cents than to cater the conversation with aeronautical terms. More and more we must train young men for aircraft and aviation sales work, not by putting them through flying schools and ground schools, but by organizing aviation business schools which will make it possible for these salesmen to intelligently discuss the economics of using the air as a medium of travel and commercial transportation.

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### Talking Safety

THE AIRCRAFT INDUSTRY'S problem is one of persuasion. Neither the manufacturers of aircraft nor the operators of air lines can remain content with their present patronage. Before we can consider that a sound footing has been established we shall have to multiply our potential clientele several times over.

There are today many thousands of people who do not travel by air, notwithstanding the possession of ample means and ample leisure. They are in a financial position not only to travel upon the air lines, but in many cases even to buy planes of their own. Their time is valuable, and they spend much of it upon the road. Still they stay out of the air. Why? That is the question which we must answer.

There may be many partial answers, but the factor which certainly plays the leading part is an enormous number of cases in fear. Some millions of Americans are still afraid to make regular use of the seaplane as an ordinary vehicle of transport, and we shall do ourselves not a particle of good by thinking that fact or seeking to eliminate it from consideration by saying that all those so affected are fools or cowards, or that their state of mind is to be blamed on the newspapers or on some other third party. The attitude of those who ought to use aviation but do not is a fact. As such we must recognize it. As such we must vigorously contend with it.

The reputation of cowardice to those who, because of lack of accurate information, or for any other reason do not agree with it that the airplane is safe enough

for ordinary employment without thought of hazard will set us nowhere. Those who do not feel disposed to providing the air line cannot be blamed or brow beaten to doing so. Every intelligent man knows somewhere the line between what he considers justifiable and unprofitable risks. Each of us will do certain things in a dire emergency that we would not consider proper as a matter of daily routine. We differ among ourselves only in the point at which our lives are drawn. What we have to do on behalf of aviation is to persuade, by precept and example, by sober argument and by emotional appeal, the great body of our fellow citizens that their individual lives honoring the attitudes which they consider proper for themselves should be drawn to include aviation.

We do violence to what many enthusiasts have reverenced as a basic article of their creed that safety should never be mentioned. They feel that it should be taken for granted, and that we should treat the prospective passenger as though no mention of that score could ever enter their minds. Despite our respect for the estimate and experience of those who hold that view, we are quite sure that they are actually wrong. We cannot suppress this question by erasing it. It seems to us rather that, if all discussion of safety is refused, the general non-aeronautical public will derive the impression that we have an inflexible complex on the subject and that the aeronautical people do not dare to talk of their safety record at length and in detail because they are ashamed of it. We do not for a moment deny that any such suspicion is justified, but the suspicion should be given no opportunity of arising.

It would seem to be a rather sensible idea if we refrained from talking so much about the crash test phrase, that the general public is amused. It is far from that, and it will take much educational work on the part of the entire American aeronautical industry before that pleasant and profitable state will be attained. There are two kinds of flying, safe and unsafe. At present the safe in the street can not tell the difference, and therefore regards all flying activity with skepticism. To educate him, and the members of his family, as to what safe flying is and to its everyday value is the major portion of our task.

This subject is too large to be covered in a single editorial. We shall return to it in some form and in its various details again and again, for we are convinced that it is vitally important. We believe that the safety of flying under proper conditions should be argued loudly. We believe that safety should be a central feature in aeronautical advertising and publicity, and that the facts should be plainly stated. Some of the propaganda which has gone out in the past has been so crude and obviously misleading as to insult the intelligence of an intelligent reader. To distribute that sort of stuff is worse than to shun and evade the issue altogether, but there is a middle ground of intelligent and fearless discussion that is far better than either.



# Tail SPINS

By CAPTAIN HARRY A. SUTTON  
U. S. Navy

Figures A, B, C and D show progress of descents conducted on the U. S. A. Liberator on the above model.

**T**HE PROBLEM of designing an airplane which will be efficient in normal operation and safe under the abnormal condition of a tail spin has recently increased in importance within the last year alone, when a great many new airplane designs have been produced. Engineering has been rushed in order to permit early possession of new models, with the result that, in many instances, serious difficulty has been encountered in meeting safety requirements. Sufficient information on airplane spinning has been published to enable a good designer to produce an airplane having satisfactory spinning characteristics of this is the primary object. The problem becomes complicated when the aim is to produce an airplane which will be as efficient during cruising (passenger or mail carrying airplanes) and which will recover from tail spins automatically, because com-

promises must be made in the design and their effect cannot always be predicted without extensive engineering investigation and experiment.

Studies made over a period of several years have demonstrated that there are several important factors which must be considered, principally, the effects of general control characteristics of the airplane, the longitudinal position of its center of gravity, its mass distribution, the variation of its lift characteristics near the angle of maximum lift, and stagger or overhang in a biplane. It was concluded at one time that compensations were not as successful in spinning difficulties as before because of the absence of control reinforcement effects between the wings of a biplane, however this has proved incorrect in several instances. There are undoubtedly differences between monoplanes and biplanes in their

bearing upon the subject. Captain Harry A. Sutton, lately of the Material Division of the Air Corps, now of the technical staff of the Aviation Corporation, will be recalled as having received the Distinguished Flying Cross, and also the MacKay Trophy for the next notable achievement of the year by a military aviator, for his work upon a military observation airplane which had shown dangerous spinning characteristics. He directed the experiments and made the test flights himself.

The subject of spinning characteristics of both military and commercial airplanes is of enormous interest at the present time. The analysis of these abnormal spins from which recovery is difficult or impossible is particularly important. It is, therefore, with special gratification that we present this article and the one immediately following, giving two different points of view, both based upon long practical experience in spinning tests and also in the derivation and application of theories



FIG. A



FIG. B

aerodynamic motions during a spin, but these differences are not necessarily controlling factors in determining the character of tail spins. Additional factors of generally lesser importance are gyroscopic torque, vertical position of center of gravity, engine operation, vertical fin surface distribution, shape and relative position of vertical and horizontal tail surfaces.

**R**esearch which will produce results generally applicable to all cases is made difficult by the numerous factors of wide differing character which must be investigated, each through a considerable range of possible influences, separately and in combination with the others. The actual motions of an airplane spin are not steady in a mathematical sense and are practically impossible to duplicate in a wind tunnel. By placing certain restrictions on the motion of a model, valuable information on autorotation has been secured in wind tunnel tests; however, these restrictions may be the cause of drawing erroneous conclusions. Any turbulent condition of air flow in model testing has always caused difficulty in applying the results to full scale performance, and autorotation tests magnify this uncertainty since the airflow must be very turbulent near a rotating model in a tested spinning state, in which models were released with an initial spin and allowed to fall freely for a considerable distance, produced some information of value, but the conditions of such tests are hard to control. It is difficult to provide sufficient height of free fall in undisturbed air and to prevent change in, or change in alignment of the model. The effects of spin distribution are very difficult to simulate in model testing, while the effects of gyroscopic torque and propeller slipstream are practically impossible to duplicate in small scale. Model testing may mean of a wind tunnel can be particularly valuable in determining relative efficiencies of various forms and shapes of control surfaces, and in developing an efficient airfoil which will have less tendency to autorotate than those now generally used.

The best design criteria now available for use in producing an airplane having safe spinning qualities were developed from study of full scale spins made on numerous airplanes of different types combined with the application of previously known principles of aerodynamics and mechanics. In the further development of basic principles it is highly important that as much as possible of the experience gained in full scale spinning be made available for general study. Actual determination of the motions of spinning in full scale is exceedingly difficult; however, the continuous measurement of accelerations and pressure distributions at various points on the airplane during a spin is practicable and would provide valuable information. Tests of this nature are generally beyond the scope of commercial organizations and governmental research agencies must be relied upon for such investigations. The study of tail spin in progress under direction of the N.A.C.A. should produce much information of practical value.

The lift of an aerial curve in a fairly uniform manner with increase in angle of attack until near the angle of maximum lift, where the rate of increase in lift begins to drop off and the maximum is reached, when a small increase in angle of attack causes a considerable reduction in lift resulting from a change in air flow conditions about the wing. This is a fundamental characteristic of all airfoils. In a normal straight fall in which the airplane does not turn or bank, the result is a gradual settling or a slow stall, depending on the elevator control available and the manner in which it is used. All pilots are familiar with this phenomenon. If the airplane is allowed to turn or roll while in a stalled attitude, the resulting motion causes a change in angle of attack and lift distribution along the wing span. The angle of attack is increased beyond the stalling angle on the downwind moving wing and is decreased on the upwind moving wing. This causes a decrease in lift on the low wing and a relative increase on the high wing which tends to increase the initial roll. The decrease in lift beyond the

roll on the downward moving wing is accompanied by an increase in drag which tends to turn the airplane in the direction of the roll. The effects of a turn while the airplane is stalled are similar, and no less long as the airplane is held in a stalled attitude; any initial roll or turn increases automatically resulting in a spin. Most pilots are familiar with these characteristics and have found by experience the difficulty of stopping a roll or turn which has been started during a stall without increasing flying speed by decreasing the angle of attack. Most of them have also found that a spin is relatively easy to stop during its initial stages. As the spin gathers momentum, other factors, such as inertia forces, gyroscopic effects, etc., enter in to make recovery more difficult. These depend largely on the speed of rotation and the problem of recovery from spins can be considered as one of providing mass means for stopping the rotation or of reducing the effective angle of attack to a point below the stall.

Considering in detail the various factors which determine the manner in which an airplane spins and its ease of recovery from a spin, it is assumed that the controls be effective and easy to operate. Aileron control is the least important of the three controls after a spin has started, but poorly designed aileron controls are often responsible for inadvertent spins. This is due to young mechanics caused by their operation in maintaining lateral level, especially at low speed and high angle of attack. If a wing starts dropping at slow air speed and the ailerons are used to bring it up, the drag of the downward moving aileron should be balanced by that of the upward moving aileron on the opposite wing so there will be no tendency for the airplane to turn. In many cases, the downward moving aileron exerts much greater drag than the one which moves up and the airplane tends to turn in the direction of the low wing. The use of rudder to prevent this often results in a stall which compels stalling the airplane and a spin results. This adverse yawing moment due to ailerons can be eliminated by any one

of several well known methods, one of which is by use of differential aileron motion where the upward moving aileron moves through a greater angle than the one which is moving down. In general, the aileron is the first of the controls to lose effectiveness during a stall, and in a spin they are of small aid. Movement of the ailerons from one extreme to the other during a spin will often have a noticeable effect on the angle of bank and will give the pilot a momentary disagreeable sensation while the change is taking place, but their effect in stopping rotation is very small. If they were capable of exerting a large drag force on the wing tip some benefit would be secured, however this is contrary to the characteristics desired in aileron controls for normal flying operations.

The rudder and rudder controls are relied upon almost exclusively in recovery from spins and are generally used in combination, although in some cases it may be possible to stop a spin by use of either the rudder or elevator alone without regard to the position of the other control. The principal effect of the elevator is in controlling the attitude in angle of attack of the airplane. In entering a tail spin it is pulled up to stall the airplane and held there to keep it stalled. If the airplane is allowed to turn while thus completely stalled, either by use of the rudder or because of dropping a wing, the usual result will be a spin. A spin can often be started by use of the elevator alone with the rudder held in neutral position, while it is positively impossible to start a spin by use of the rudder alone if the elevator is left free or held in neutral. After the airplane has started spinning, the rotation sets up gyroscopic forces on the wings and, usually, inertia forces which tend to maintain a stalled attitude, and the spin will often continue if the elevator is released. The safe spin is one in which it is necessary to hold the elevator up to maintain the spin after in this case the airplane possesses an inherent tendency to dive out of the spin and will do so if the controls are released. As the spins continue,

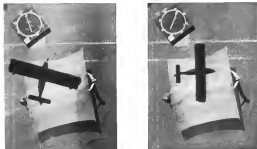


FIG. 1

FIG. 2

AVIATION  
December 31, 1939

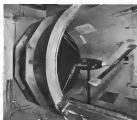
AVIATION  
December 31, 1939

if the elevator load is reversed and pressure is necessary to prevent the slide from coming back, the aileron will not stop spinning if the controls are freed and it may be necessary to exert considerable pressure on the stick to change the attitude sufficiently to stop the spin. In such cases the airplane will generally spin at a progressively faster attitude if the elevator is left free until the force needed to decrease the angle of attack sufficiently to stop the spin may exceed that capable of being exerted by the elevator. Depending more than be placed on the rudder to stop or slow up the rotation. The rudder is usually less effective in stopping a spin than the elevator. Stalling in the primary cause of spins and spins are controlled by the elevator while the rudder exerts its effect in recovery from a spin in a secondary manner through downing up or stopping the rotation, after which recovery from the stall must be effected by the diving tendency of the airplane or by use of the elevator. It is to be expected in the normal case that the elevator will be more effective than the rudder in a spin. Both are generally used in entering a spin and in the recovery.

As the attitude of the airplane and its speed of rotation change during a spin, the airflow conditions at the tail surfaces are modified and the effectiveness of the elevator and rudder may be greatly reduced. Tail surfaces are designed to be more effective under normal conditions when the air stream strikes them from the front at relatively small angles. As a spin the attitude of the airplane and its motions of rotation continue to greatly change the direction from which the air stream strikes the tail surfaces, so that it should not be surprising to find that in some cases a particular control which is normally sufficiently effective is almost totally lacking in effect. Usually only one of the tail controls is affected in this manner and in most instances the elevator retains effectiveness although the loads on it may be very high. No satisfactory design criterion exists for avoiding this condition since no exact data is available regarding its cause. It would seem best, however, to avoid rectangular stabilizers of large chord and radius of curvature behind the stabilizer. Fig. 1 shows possible banking effects. This is probably the reason that in some cases there is practically no air pressure on the rudder during a spin.

THE ANGULAR rates of elevators should be properly proportioned so that the upward motion permitted is only slightly greater than the downward motion. Each should be equally efficient for satisfactory control in normal flying and in recoveries under the various load conditions, and the stick control motion should be such that a normal pilot can, without reaching, place the stick in any permissible position, particularly with the stick in its most forward position. The elevator control force should be small and the change in elevator force required for balance between power on and power off conditions should not be excessive. The elevator should retain effectiveness to the stalling point, while the rudder should be the last of the three controls to lose effectiveness in a stall and should have ample power at slow air speed without requiring excessive control force at any normal air speed. The rudder controls should be capable of being reached by a normal pilot with the rudder in either extreme position. Both control systems should be positive without appreciable slack or stretch.

Longitudinal position of the center of gravity affects spinning characteristics in at least two ways: through its



Which barrel after the rotating axle made at K.A.C.A. International, Los Angeles, Cal.

effect on inherent longitudinal stability of the airplane, and by the relative change caused in mass distribution and inertia moments. An airplane which is made very stable by placing the center of gravity forward of its usual position is difficult to spin, due to the lack of sufficient elevator control to properly stall it, and to the tendency of the airplane to be very easy heavy during a spin. These effects are distinct advantages from the standpoint of safety from inadvertent spins provided sufficient elevator control is provided for satisfactory normal flying maneuvers. The usual military airplane, balanced between 30 and 35 per cent of the mean aerodynamic chord and the longitudinal controls are designed to provide sufficient effectiveness for stalling, making three-point landings, and getting the tail up during take-off in this range of balance location. If the balance is such an airplane is moved to 25 per cent of the M.A.C., it is generally necessary to provide more effective longitudinal control in order to stall the airplane or make satisfactory landings. A majority of the commercial airplane types have been so designed that the balance location with full load is between 35 and 40 per cent of the M.A.C. and in the cabin types particularly there is usually a large change in balance location between the full load and empty conditions. This is undesirable both from the standpoint of control and longitudinal stability and has undoubtedly resulted from the fact that more attention has been paid recently to looking than designing airplanes. It requires careful study to design a cabin type airplane in such a manner that the actual balance location will be satisfactory and as the change in balance with various load conditions will not be excessive, however, sufficient information will not be excessive, it is essential that it be applied to airplanes are to be made safe and otherwise satisfactory for commercial use. The usual airplane requires that the balance be not further back than 35 per cent of the M.A.C. for satisfactory longitudinal stability, and for safe spinning it is highly desirable that it be nearer 25 per cent. Much has been written on the disadvantages of too much stability, but these were more imaginary than real so far as longitudinal stability is concerned, and it is certain that many of the commercial airplanes recently designed in this



rotation about the axis  $A-C$ , there will be a tendency for  $Z-Z$  to rotate about it as perpendicular to the axis of rotation. This will occur unless  $Z-Z$  is able to coincide with the axis of rotation  $A-C$ , but any slight displacement from this position will cause it to swing to a position at 90 deg. to  $A-C$ . The condition is thus one of unstable equilibrium. If the moments of inertia, wing reversal,  $I_{Z-Z}$  would tend to rotate until it reached a position at 90 deg. to  $A-C$ . This applies in principle to the spinning airplane, although the axis of spin is not generally through the center of gravity but is at some intermediate and variable location on the inside of the spin, which makes calculation of the dynamic couple only very roughly correct. In the usual airplane  $I_x$  will be larger than  $I_z$  and a dynamic couple will be created which will cause positive rotation or an increase in rolling attitude. It will be noted in the experimental determination of  $I_x$  and  $I_z$  given above that there is a wide difference in the staggered location but an appreciable difference in the unspinning airplane, especially if the calculated values are used. This may have accounted for the difference in their attitudes in a spin.

To assist in measuring safe spinning qualities, moments of inertia should be carefully calculated by making the airplane complete; and particularly the difference between  $I_x$  and  $I_z$  should be small since the magnitude of inertia rolling moments is determined by this difference, the attitude of the airplane, and the rate of rotation. In practice, any airplane in which the control forces are not in excess of the control forces, that is, where the pull on the control stick becomes zero and a forward pressure is necessary to keep the stick from coming off the way back, should be brought out of the spin immediately. It should be noted that the effects of center of gravity location and inertia moments must be considered together since changes in center of gravity location are usually made by shifting loads in the fuselage which also changes the moments of inertia.

IT IS WELL KNOWN that the fundamental cause of tail spin lies in the possibility of auto-rotation. The comparatively sudden reduction in lift here, or, more exactly, normal force on a wing past beyond its angle of maximum lift, is responsible for the possibility of a rotation being sustained after the engine has been started. This fact has been repeatedly demonstrated both in theory and experiment and it has been shown that changes in shape of the lift curve near its maximum have a marked effect on auto-rotation. If instead of abruptly falling off beyond the maximum, the lift curve could be made to remain at approximately its maximum through a considerable range of angle of attack followed by a gradual reduction in lift forces as on Fig. 28, the probability of spinning would be greatly reduced. The unusually large angle of attack necessary for auto-rotation under such conditions would lead to elevator spin in the normal airplane, since it would not have sufficient longitudinal control to reach such a large angle of attack. However, if a spin did get started it would probably be very difficult to stop due to the abnormal attitude. Most of the research on airfoils has been devoted toward obtaining a higher maximum lift with low drag and the most efficient airfoils exhibit a sharp peak in the lift curve. Various combinations of airfoils can be made to modify this feature but no airfoil or combination of airfoils has as yet been developed which could high lift, low drag, and the absence of a peak in the lift curve. A few airfoils using airfoils with comparatively flat top

lift curves have been built as a sacrifice in aerodynamic efficiency. Others have been built to accomplish the same purpose by providing longitudinal controls sufficient to permit stalling the airfoil; however, this is obviously dangerous practice. Modification of wings by the use of wash out at the wing tips or by changes in plan form are intended to delay stalling of the outer portions of the wings, benefit mainly by decreasing the probability of inadvertent spinning and decreasing the rate of rotation during a spin.

THE USE of auxiliary surfaces or controls such as flaps and slots has been advocated as a means for spin recovery difficulties, and a large amount of experimental development has been carried on, particularly in England where a large number of airplanes have been equipped with automatic leading edge slots. These are operated by the air force on the wing as such a manner that at large angles of attack they are opened thereby delaying the breakdown of smooth airflow over the wing and increasing the lift. At low angles of attack they are closed and conform to the contour of the airfoil section used as aerodynamic characteristics in normal angles of attack are practically unaffected. In a spin the wing opening operates at a higher angle of attack than the outer and the slot on the inner wing is prevented to increase the lift and assist in recovery. Actually the angles of attack are sufficiently high on both wings that generally both slots are open during a spin. In the normal stages of starting a spin these slots do assist in preventing the airplane from rolling and provide increased lateral stability and control, thus increasing ability to some extent by generative means. By incorporating a leading device which enables the pilot to bring into action only the slot on the inner wing during a spin, some aid in recovery may be expected, although it must be remembered that the spin also increases the wing drag. The development has not yet reached the point where a designer can be assured of correcting spinning difficulties by the use of slots.

The use of auxiliary devices intended to provide increased lift or drag on parts of the airplane where the results would be beneficial in recovery from spins is not desirable unless they can be made entirely automatic in action, as the pilot is unable to operate them or use additional controls during a spin. It is to be hoped that no further additions will be necessary to the mechanism of an airplane as most of them are already complicated at present.

Present knowledge of spinning indicates that adherence to sound fundamentals in airplane design and the application of a few general principles should produce satisfactory results. The center of gravity should be kept well forward, ahead of 30 per cent of the mean aerodynamic chord. This will also assist in increasing proper longitudinal stability. The moments of inertia should be kept small, particularly the difference between moments of inertia about the vertical and longitudinal axes. All of the controls should be effective and easily operated under all normal conditions and especially at large angles of attack. Designer should be used as a highly and possible interference effect between horizontal and vertical tail surfaces in a spin should be considered in determining their shape and relative location. It must be admitted that application of existing knowledge will increase some element of doubt which only well reasoned research will remove. Every aircraft designer should be given complete research organizations engaged in solving this problem.

## THE Flat SPIN

By LIEUT. RALPH A. OFSTIE  
U. S. Navy

AS A general proposition it is true that any airplane which can be sharply stalled can also be spun. Further it may be readily matched that a large number of the smaller military and commercial aircraft which have been recently flown can be made to "flat" spin, although sometimes with difficulty and under somewhat unusual conditions as to loading, balance, and control disposition.

There seems to be but little agreement as to the proper terminology for this type of spin. Some groups of pilots may be referred to as the inverted or inverted spins, autorotation, stable spin, or autorotational spin. All of these, however, refer to one and the same thing. "Flat" spin has been selected to cover the one termite it is perhaps the most correctly used although very few flat spins are, strictly speaking, actually flat, nor do they even approach that attitude.

It has been recent newspaper practice to refer to the "deadly" flat spin as though it were something quite new as the attitude however, and a characteristic which immediately conditions the attitude in which it is assumed. It is quite true that a goodly number of pilots have lately been killed by crashing from prolonged spins which obviously were uncontrolled. But it is entirely probable that the flat spin has been with us just about as long as the conventional spin of years gone by, and that merely advancing knowledge—and more spinning—have brought up the recent determination of exact phenomena which differentiate the flat spin from the ordinary garden variety. In any case the flat spin is definitely with us, and an endorser is made in this paper to present some practical information relative thereto which has resulted from somewhat extended spinning experience, and to offer certain opinions of the present general attitude on the subject.

The exact degrees of entry into a flat spin vary rather widely between types, and may be somewhat different at various two phases of the same type and loading. Generally speaking, however, the entry occurs as follows: the plane is stalled as far as normal spin, usually at a level stall, but sometimes rather deeply. Pull up elevator, and full rollout in turning direction are released thereafter. The plane is held in the conventional spin, normal for the type, for four to six or more turns, tending to move them toward the normal position. Then follows a brief period of transition, perhaps one

The present paper, by Lieutenant Ofstie and the preceding one by Captain Sutton, admirably implement each other, having been based upon studies of different types of planes and different methods of testing, and also treating the subject from somewhat different points of view. Captain Sutton dwelt especially upon the general theory of spinning and the way in which changes in design affect spinning characteristics. Lieutenant Ofstie concerns himself particularly with the practical operation of planes with abnormal spinning characteristics, and with the general phase of the spin in present day aerial operation. Lieutenant Ofstie, now attached to the flight test section at the Naval Air Station at Ansonia, has had a varied experience as a naval aviator.

term, after which it will be observed that the rate of the dip has eased somewhat from a very small amount up to as much as 60 deg., the latter one being probably almost at the horizon. The spin rate is lower than the outside, a peculiar whirling sound is noted, caused by the air stream striking streamlined wings and exposed objects at unusual angles; the engine vibrates as though the propeller were out of balance; the pilot may feel himself pushed against the back of the seat by centrifugal force, and finally, the control forces are reversed—up is down, down is up, and they are all right and in addition the inward alone (now moving backwards) takes charge to the extent of pulling the stick to the inward corner. This is the "flat" spin, and unless some radical sense is taken the spin will probably remain flat and perfectly regular in its rotation, until the ground comes up and stops the whole proceeding.

ACTUALLY, what has happened? To start with, in the normal spin, the plane has been held in a stall by the elevator, over which a normal airflow has been passing. At the same time, the plane has rotated about its axis well beyond its normal angle of lateral descent with the wing surfaces at an average angle of attack only slightly beyond the burble point of the air foil. Recovery from this condition requires merely that the stall be discontinued (elevator moved down to reduce angle of attack and permit greater forward speed), and then the turning movement be stopped with the rudder. Both of these controls are quickly effective because they strike the airstream as a normal stream, except at a somewhat reduced air speed. But all these conditions change when the flat spin takes charge. The control of rotation has moved back to a point within the airplane, usually near the center of gravity; the angle of attack of the wings may now range from a small positive angle to the outer

up to 90 deg. near the center section and nearly 180 deg. at the main wing tip (that is, the main wing is moving backward with reference to the surrounding air), the elevators and rudder, instead of being held flat against a normal airflow by the pilot, are now held against the wake by an unusual airflow; the engine thrust line, instead of being reasonably near the flight path, is now



The last turn of a flat spin. Fig. 3

almost at right angles to it, and the propeller spins to do its worst at almost abnormal angles. These conditions all combine to cause the peculiar characteristics which the pilot recognizes as the flat spin.

The flat spin sounds rather bad, and sometimes it is. Certainly the first experience, at least in some types of planes, is decidedly disconcerting. But unfortunately the usual recovery follows the book for normal recovery, slightly emphasized: push the stick in the dash, with full opposite aileron and rudder. The "almost" of the flat spinners will recover in slightly less than two turns, some wait about three, a few require a half dozen, and an occasional one simply will not respond to this treatment. The latter case is discussed later; the rule which holds in the very large majority of types, however, is to push the stick in to the center, back the rudder hard over, and hold them there.

To show the relatively wide variations possible in the spinning characteristics of different types the following examples are cited: the planes in all cases being lightjanes, carrying full load, and all recovering from their normal spins in about a full turn.

(a) Two-place, high performance. Plane enters a normal spin quite readily. Makes about five extremely fast and regular turns with nose almost vertical, then slows down appreciably with nose rising about 20 deg. The normal spin is faster and much less uncomfortable than the latter flat spin; recovery, however, requires two full turns from the latter condition.

(b) Training plane, 300 hp. class. Plane enters a normal spin with some difficulty. The first three turns are slow and softening, with a distinct wisp (about once per turn) which unconsciously raises the nose of the wings. After about the third or fourth wisp the nose stays high, perhaps 30 deg. below the horizontal, and the speed of rotation almost doubles. Recovery takes a full three turns.

(c) Single seater, high performance. Apparently will

not flat spin unless stabilizer is adjusted full tail away. In the latter condition the nose quickly comes up to about 30 deg. from the vertical. Recovery very difficult, effected in two cases by pilots attempting to leave ship (with parachutes) and, on moving their bodies forward and up, finding the ship came out on a dive.

(d) Light two-seater, 100 hp. Spins easily, but with nose almost vertical. At about the fourth turn nose comes up about 30 deg., and flat spin is fully developed in one additional turn. Recovery is made in two turns.

(e) A training plane with 200 hp. engine. As a hand plane with stabilizer set at a small dihedral angle, this craft spins with some difficulty. Once started, however, it goes into a flat spin after about three fast turns, and recovers readily in less than two turns. As a surprise, with the nose dihedral in the stabilizer, recovery takes three turns, but with stabilizer set for zero dihedral the recovery requires about six turns.

The few examples cited above illustrate the rather wide variations in entry, attitude during spin, and time of recovery, that may be expected in a fairly diversified group of planes. But once the spins are actually started, whether the nose be high or low, the same conditions obtain for all: loss of thrust, reversal of control forces, and markedly delayed recovery as compared with the normal spin.

A number of types seem to be borderline cases. A small increased angle of the elevators, a few degrees negative stabilizer, a few more pounds of load back of the center of gravity, and an apparently perfect airplane suddenly becomes a flat spinner. Conversely, by setting the stabilizer positive, by keeping the load well forward,



Fig. 4

and by easing up on the stick just a small amount, an otherwise regular flat spinner can be demonstrated to have none of the characteristics of this class. The margin is surprisingly small in many cases.

The military air service permit prolonged spinning in several classes of small planes, with restrictions against certain specific types in which dangerous tendencies have been encountered. Some of the planes in service, particularly of the training types, have been so modified after their aerial appearance that had spinning characteristics have been practically eliminated—at a cost, however, of some of the control and maneuverability, and with limitations on loading. The flat spin is definitely recognized, and, for all practical purposes, is accepted as bad news which must be taken with the good.

The regulations of the Department of Commerce divide somewhat sharply from those of the military service. Their present requirements apply to sport cockpit planes of less than 4,000 lb. and sales made of slightly less weight. Briefly they insist that planes in these categories shall recover from a six-turn spin in one and



Fig. 5

one-half turn with normal controls, no power, and no reversal of stick forces. While it is undoubtedly desirable that the winning characteristics should meet these requirements, yet it is questionable if a smaller set of approved recreational planes honestly do so.

As has been previously stated it is not difficult to present a plane for test which due to a slight loosening on elevator throw, and moderate shift of axial loading will be incapable of entering a flat spin. Furthermore the pilot can without much difficulty avoid the flat condition by easing slightly on the stick, particularly in planes which have a pronounced "indicator" (wobbling of control stick) just before going flat. The point is it is desired to stress it that a plane type might pass the Department of Commerce test but after getting out into flying service might develop so-called undesirable spinning characteristics, caused by slight change of load and control arrangements.

But after all, why should the flat spin be so heartily and generally condemned? Deliberate prolonged spinning is almost universally regarded before the airplane will go "flat." Just what purpose it is desired to accomplish, just what value the prolonged spin has, is



Fig. 6

difficult to see. Obviously every student pilot should have some spin training, but one or two corrective turns are certainly sufficient to illustrate adequately the working of the spin and the recovery therefrom. These are deliberate spins, entered into at a safe altitude and with the pilot adequately warned as to what is about to happen. The characteristics of the spin are known; the student has been carefully instructed and checked. He spins in order to learn what to do instinctively in case of an involuntary spin.

Many so-called spin facilities are improperly placed in that category, that is, spins which originate close to the ground. Actually the spin is merely outlasted falls off to one side with perhaps a half turn in that direction, and goes in on its nose before being stabilized in a normal spin. Reduction of fatalities from this source involves more careful training in appreciation of the stall condition, and immediate disavowal of controls for recovery of flying speed. The point is that the plane does not even start spinning, and therefore this type of crash can have no relation to the flat spin.

THE UNWILLINGLY SPIN AT altitude may result from any maneuver which leaves the plane in a stalled attitude. It may occur while maneuvering, practicing turns or other maneuvers, or it may be caused by weather conditions in which a pilot, temporarily flying blind, has lost his sense of direction and ground relation. But once the plane has been stalled unvoluntarily, and falls to a spin, a qualified pilot should quickly recognize the condition and take prompt means for recovery. In the case of thick weather flying the pilot may do a successful dive,

spends and spins; but each time the spin is recognized he must ease his stick forward for recovery, and each time a dive is apparent (evidenced by high air speed) he eases his stick back. Of course it is not quite as simple as that, but unless the pilot is thoroughly experienced he would hardly be out attempting blind flying in dangerous weather. In any case no pilot will permit himself to stay in an involuntary spin much over a turn, and certainly, when the flat spin can be corrected by merely easing the stick forward he should encounter little if any trouble from that source.

Some pilots must make prolonged spins in the testing of new types of aircraft, and particularly training planes. Since it must be done, and necessarily has a definite element of danger, the testing should follow a course which will limit that danger and yet permit the correct data being obtained. The test pilot should first see that his plane is tested exactly in accordance with production specifications. This applies particularly to the angle of slaps of the skidstrut and rudder. The plane should

Two spins should be entered at an altitude certainly on the high side of 8000 ft., and once the plane has finished half a dozen turns of a normal spin, or a spiral for a few turns in a steady flat spin, recovery should be made. The normal recovery has already been discussed if this proves ineffective after about six turns more force should be exerted to give motion in the flying direction. Should this be ineffective after several additional turns an attempt should be made to rack the plane out, using the engine in conjunction each time the controls are moved for recovery. It is necessary, of course, to work with the natural period of the plane in attempting recovery by this means, the controls being operated very much in the same manner that a car is rocked on the step for takeoff. But if all the above recovery attempts prove useless the dump valve on the hopper is pulled, assuming that the plane had reached a loaded condition when trouble was encountered. This last should result in a movement forward of the center of gravity sufficient to permit normal recovery. If not, remember what paragraphs are made for

Assuming that the plane has been prepared as previously referred, the following specific procedure is suggested in the spin testing of new types of planes. The same tests for speed, climb, stability, etc., will have been previously completed.

The first flight is made with center of gravity as the forward as possible, that is, with pilot in front seat and no ballast in rear. Set stabilizer for inherent cruising speed. Pull the plane without power using maximum rudder and elevator. If plane goes into a simple spiral and refuses to spin, try again with full opposite ailerons and negative stabilizer and a burst of the gas each time the plane whips. If none of these is effective it is assumed that the center of gravity is too far forward to allow of entering the plane in a spiral.

The next flight should be made under the same conditions as the first except that an increment of shot lead should be put in the hopper. On subsequent flights this lead is gradually increased until the hopper is filled, and finally the full specified load is carried. In the latter case if the flat spin has not yet developed an effect should be made to induce the condition as before by using full opposite stabilizer and bursts of the elevator.

Finally if the plane does spin flat and if it has dual control, a last test should be made to determine whether recovery can be properly effected using the rear controls. For this purpose the shot hopper is removed; pilots flying in both seats with the forward seat acting as safety pilot to take control in case the centrifugal force in the rear is too great to allow proper use of the controls from

flat spinning is uncomfortable, uninteresting, and frequently unobtainable. It is a necessary preliminary for training operations. One can keep one's head in on spins by very occasional practice—not more than a couple of turns consecutively—as a training plane whose characteristics have been carefully determined. Repeat that as a golden rule, adopt the policy of never spinning voluntarily. It gets you nothing except some day perhaps, a good crack-up.

## THE Dornier Do.X

### A Flight Experience and Some Remarks on the Operation

By LIKUT. COMMANDER. JAMES M. SHOEMAKER  
U. S. Navy

SEPTEMBER fourth, 1929, was a good day for the little town of Friedrichshafen on Lake Constance. In the morning the Graf Zeppelin landed unexpectedly up the Rhine and across the lake to its home harbor, where it was met and applauded by thousands of proud and happy Germans headed by members of the Reich cabinet. The ambassadors from the United States and from Japan were

on hand to take part in the welcome and to receive, on behalf of their respective countries, expressions of gratitude for the assistance afforded Dr. Eckener and his crew in the world-flight just completed.

After the crew and passengers of the Graf Zeppelin

*Performance records of the Do.X have been widely published, and many journalists have written extensive impressions of the big flying boat. Lieutenant Commander Shoemaker, however, tells of his reaction to a flight over Lake Constance from the point of view of an expert, having been in Germany as official observer for the U. S. Navy on board the Graf Zeppelin. When he writes of the performance of the Do.X, he is comparing it with a broad, scientific experience in all types and varieties of aircraft.*

had been deposited at the Kurgarten Hotel, and were sending their first greetings for three days, a gradually-increasing line of many others was heard to the south-west. In a few moments Dr. Dornier's latest product, the twelve-engine Dornier Do.X, swung overhead and glided down in a landing in the lake.

That same afternoon, the three American naval officers who had been guests aboard the

Graf Zeppelin accompanied Ambassador Schoenbrunn and the tremendous Do.X. Two motor boats carried about fifty people out to the plane, which had been waiting at her mooring off the town. As we approached the plane, the



The Do.X—before the crash. It will be noted that the other two descended near a short point throughout the test.

be flown from the most forward of the control cockpits in order that centrifugal effect will be a maximum if a flat spin is encountered, the reason being that most ships are about as soon passing approximately through the forward seat in the conventional double cockpit airplane. Finally a hopper for shot ballast should be rigged in the rear (cockpit) in which variable loads of small shot can be placed during the progress of testing. The hopper is fitted with a suitable check valve with an operating handle close to the pilot.



The plane describes its right-hand circle, and then circles



# Mapping 13,000 SQUARE MILES OF Alaska

By LIEUT. E. F. BURKETT, U. S. N.

*Executive Officer, Alaska Mapping Expedition*

**M**APPING of 13,000 sq. miles of northeast Alaska, not including 649 oblique photographs, tell part of the story of the 1929 Alaska Aerial Survey Detachment from Aircraft Squadron, Battle Fleet. The survey covered principally the Tugena National Forest of which 4,000 sq. miles remain unmapped.

The photographic results of the expedition were 42 rolls of tri-lens mapping film, making a total of over 25,000 single photographs covering the area. The films have been delivered to the U. S. Geological Survey where they are filed for future use. In addition to these the negatives of 452 oblique photographs were turned over to the Forest Service. The Bureau of Aeronautics has on file the remaining negatives.

In addition to the value of these photographs in map making, they have proved of considerable value to the Forest Service. During the last two years great trouble during the 1926 expedition have been in the office of the Forest Service at Juneau, whose officials have realized they are of great value in estimating the area of the forests and to a large extent the quantity of timber on the areas over which the permits were taken.

"It is impossible to estimate, in dollars and cents, the value of the results of the expedition," commented R. H. Surges, representative of the Department of Interior and Agriculture, and former member of the expedition. "In fact, it will be years before the total worth of the results can be determined. Only those who have experienced the difficulties of conducting topographic mapping by ground methods in such a country as northeastern Alaska can appreciate the invaluable aid of the mapping pictures."

In this country, where climatic conditions are almost at their worst, topographic mapping is slow, arduous and costly, not only because of the inclement weather but also because of the heavy blanket of timber and almost impenetrable underbrush. Accurate topographic mapping by ground method is well-nigh impossible at least within the bounds of wintered cost. By such methods, the topography of the shore lines and mountain ridges can be obtained with comparative ease, but accurate discovery and mapping of the valley bottoms with the many lakes and stream streams is prohibitive. For this purpose the aerial photographs are available and will serve to make the maps of the islands and

*Although the 1929 Alaska Mapping Expedition received little or no publicity outside of governmental and commercial aeronautic circles, it was without question one of the outstanding aviation enterprises of the year. During the time 677 40 hours spent in the air by four Navy Wasp equipped Leasing Amphibians a total of 13,000 square miles of northern territory was mapped. In this article, Lieut. E. F. Burkett, U. S. N., executive officer of the Expedition, relates in detail of the work accomplished, and of the personnel and equipment necessary to the success of such an undertaking.*

swathed of northeast Alaska much more accurate and possible of accomplishment in much less time."

But what did the survey undertake and accomplish? After the detachment was formed at San Diego and proceeded to Alaska aboard the *Gannet* and a converted barge, and to the air in four Wasp-powered Leasing Amphibians, the mapping planes flew 242 95 hours while undertaking mapping runs. Possibly the following data will be interesting:

Hours in the air for all planes, 677 40  
Miles flown by all planes, 24,182  
Gallons of gasoline used by all planes, 16,887  
Hours in the air for photographic planes, 242 95  
Miles flown for photographic mapping, 19,346

Gallons of gasoline used for photographic mapping, 8,105  
Rolls of tri-lens mapping film exposed, 42  
Number of tri-lens mapping exposures made, 7,980  
Rolls of "D" or fourth lens mapping film exposed, 10  
Number of "D" lens mapping exposures, 1,900  
Number of mapping negatives, 23,640

Number of square miles covered by photographic mapping, 13,000  
Number of press maps from mapping negatives, 600  
Hours in the air for oblique photography, 95  
Miles flown for oblique photography, 7,752  
Gallons of gasoline used for oblique photography, 1,236  
Number of aerial oblique exposures made:  
For Bureau of Aeronautics, 229  
For other Federal Bureaus, 467  
Number of prints made from oblique exposures, 5,600

**T**HE PLACES first overflown from San Diego to Seattle during its opening hours, but with few exceptions the regular steamer route via the inside passage was followed on the flight from Seattle to Kotzebue, due to the fact that it is usually over well protected water and is one of a forest landing assistance would be more readily available.

As the distance between Seattle and Kotzebue could not be covered in one flight, Alert Bay, B. C., 204 statute miles northwest of Seattle in Johnstone Strait was selected as the logical refueling point. All four planes of the detachment were based up on the beach. Although Alert Bay under present conditions is considered the most feasible stop-over point for planes flying between Seattle and Kotzebue, there are several other well protected harbors along this route where station refueling facilities are available.

Similar to general conditions in southeastern Alaska



View of the Landing Amphibians of the Alaska Mapping Expedition in a Semakine Bay.

was concerned, later experience gave a clear-cut and positive proof of the advisability of a rugged amphibious such as the *Leasing* for this work. No suitable land fields were available. Unusually deep water and an extreme tide range of 24 ft with resulting strong currents in the channels along which the most important cities are located usually made anchoring the planes impracticable.

At places where sufficient dock space was available the planes were secured on the docks, the *Gannet* towing them to and from the water as operations required. Where dock space was not available, the aircraft were beached and secured and the planes landing on the water and taxing out on the beach under their own power. It is in connection with such operations that the value of the characteristics of the amphibious type of plane is fully realized and appreciated. The best beaches were barely large enough the water coming in to the sides of the planes during the extreme high tide which occurred for a few days each month.

At Juneau, the *Gannet* and large were moored to the Government wharf and the planes secured on the wharf. This wharf was in excellent condition and equipped with telephone, water and power connections which were made available for the use of the detachment. The operating conditions were very satisfactory and permanent construction of all sorts of the detachment. The officers and crew of the *Gannet* became extremely efficient in handling the planes to and from the dock, and always gave excellent cooperation in the work to be done.



Before the summer was over, all first planes could be hoisted from the water and secured on the dock within 12 min. This became a vital factor in cutting down the interval at noon between mapping flights as the summer passed and the time limit for aerial photography was retained. Aviation gasoline and oil in 50 gal drums was delivered to the docks by the Jetsons (south of the Standard Oil Company of California).

Operating conditions at Kotzebuk compared favorably with those at Jensen and photographic work was carried



Inventory of supplies valuable for operations (these reserves

on with a minimum of lost reserves. Aviation gasoline and oil in 50 gal drums was delivered to the dock by the Kotzebuk branch of the Standard Oil Company.

During the summer in northeastern Alaska each place was given the regular 20 hour check as required by existing instructions and it was largely due to the care and thoroughness with which these were carried out that engine failures were entirely eliminated. During very weather, which frequently occurred when the three-week due for this check, a narrow boat fly was rigged over the engine and the check carried out, entering the midships of the plane for the most mapping weather.

Excellent performance was gleaned from the four Pratt Whitney air-cooled Wasp engines, and all four planes completed the summer's work without a change of engine. In view of the unfavorable operating conditions due to limited beach space, lack of shelter, rapid temperature changes and heavy rainfall it is considered that both the planes and the engine withstood the rigorous test remarkably well.

Although beset with frequent and heavy rains, the detachment was ready to commence photographic operations on the day of our arrival at Kotzebuk on May 24. Although photographic operations could have been car-

ried on from Kotzebuk, it was decided to wait until the sun was to be mapped. An aerographer, with complete equipment for taking upper air soundings and for making a daily weather map and forecast, proved his worth at this stage of the operations. Unfavorable weather for photographic flights was predicted for the next five days, but on May 29 we had reached Petersburg and were ready to operate.

FROM PETERSBURG the winds of Kila, Corcoran, Warren and the southern half of Baranof were mapped. After investigation and comparison of the facilities at Sides and Jensen it was decided to shift the detachment from Petersburg to Jensen for the remainder of the work to be done in the northern half of the Tongue National Forest. On June 17 the entire detachment moved to Jensen, and while operating from this base the northern half of Baranof and the islands of Chichagof, Iliamna and Yakuba and portions of the mainland were completed.

After operating from Jensen for two months, the detachment departed on August 17 for Kotzebuk, arriving there two days later, after stopping at Pitmeagut over Sunday in a downpour which continued without cessation for seven days. When the weather finally cleared, the detachment mapped sections of the mainland between Behm canal and Portland canal and a few scattered areas in the north along Stephens passage.

Although there was much rain during the summer, the weather usually was the same throughout the area, although occasionally we were able to locate a clear area despite apparently unfavorable conditions. With this in view, it was suggested to arrange for weather reports at least twice daily from as many local stations as possible. On one occasion while based at Jensen, four aerological mapping flights were made in the vicinity of Glacier Bay where it was cloudless, while at Jensen and to the south of the area was cloudy with occasional rain.

If the weather were suitable for mapping in the



First (left), Harold G. Smith and Lt. W. H. Smith (left to right), standing with the rolled photographs.

## AVIATION December 31, 1939

## AVIATION December 31, 1939



Officers and crew of the 1939 Alaskan Mapping Expedition at Jensen.

morning, it seldom cleared sufficiently during the day to permit mapping operations. Frequently the weather was clear enough for mapping early in the day, but clouds formed by noon. The first mapping weather after a rain was usually extremely clear, but a day later the weather, even though cloudless, became slightly hazy with the haze increasing in density with the passage of each clear day. During one stay the following eighteen days were clear enough for mapping: May 20, June 4, 5, 6, 15, 25, July 10, 11, 12, 18, August 1, 2, 3, 4, 5; September 1, 4, 6. Of our 104 days in Alaska, rain fell on 73!

Three precautions were taken to insure the safety of any mapping crew. First, of course, the route of the flight was definitely determined in advance, with alternate flight lines in case of foul weather. Second, emergency routes were provided each place to land in case in case of a forced landing at any distance from a base.

Each plane was equipped with the standard army medical aviation first aid kit, and with a stock of emergency rations in an ordinary givie pillow box, the top closed and soldered to make it entirely water tight. The rations included 3 lb. can orange, 15 pieces hardtack, 2 lb. bacon, 2 lb. dried fruit, 2 lb. coffee, 2 lb. sugar, 2 candies, salt, pepper, 2 pieces paraffin soaked paper for making fires, 3 lb. can corned beef, 1 lb. bacon, 2 lb. sweet chocolate, 2 lb. chocolate, 1 carton cigarettes, 1 tin of sulphur matches, fish hooks and line, 1 can for making coffee.

Two of the four planes carried radio equipment. In making plans for the survey some difficulty was experienced in deciding just what aircraft radio equipment was desirable and necessary. It was finally decided to equip the two SE-130B-A transmitters and two SE-240S-A receivers, and an SE-130B-B transmitter-receiver. Two of the planes were wired for radio equipment before leaving San Diego, but radio sets were not carried between San Diego and Seattle.

Only a few minor radio troubles developed. Two fans developed vibrations and were replaced while the CW 1544 tubes after being in use a short while would become garbled or lose their life. Care was taken to insure that the tubes operated at their normal filament-voltage rating according to their tip coloring, but the reason for

their going dim could not be determined. This necessitated carrying plenty of spare tubes in the plane.

An improved installation of the antenna reel and lead cut manager should be made as the present one causes too much strain on the reel. On one occasion the reel handle broke and on another the reel mount broke, thus requiring the operator to hand the antenna in by hand and pile it in the bottom of the cockpit. In fitting out another expedition for work in Alaska, the plan of taking two of the most powerful and rugged aircraft sets available should be followed. The smaller sets are almost useless.

PRACTICALLY all flights required operation in darkness well over 100 miles under atmospheric conditions generally very unfavorable. No high frequency set was available for the expedition. The excellent results obtained with the aircraft radio were due to a large measure to the efficient cooperation of the Canadian government, the U. S. Army and U. S. Navy radio stations.

While our most important operations were carried on in the air, the ground crew have been accomplished without a base. Immediately after the 1939 expedition had been authorized, the Puget Sound army field command overhauled covered large YF-86. This large was originally a standard 110 ft by 25 ft army ammunition lighter, but had been converted for use as living quarters, workshops, office, store room and photographic laboratory for the summer detail of the 1936 expedition. The large was equipped for hot and cold running water and wood for electric lights and power.

The work during the summer was performed for several governmental departments, notably the Departments of Interior and Agriculture. Lt. Col. A. W. Rutland was officer in charge. Other officers and their responsibilities were as follows: Lt. E. F. Barkan, executive officer, operations, navigation and communications; Lt. R. F. Whitehead, photographic aerological; Lt. C. F. Gabel, engineer, radio, athletic; Lt. (jg) L. P. Pavlovskoy, assistant aerologist; flight, material; Lt. C. A. C. Smith (MC), medical, welfare; Lt. B. F. Carr (SC) supply, disbursing.

# THE Goodyear-Zeppelin Dock

## AT AKRON, OHIO

By WALTER E. BURTON



Above: A close up aerial view of the south flank of the Akron Dock. Below: An air shot of the Goodyear-Zeppelin dock taken last Nov. 11.



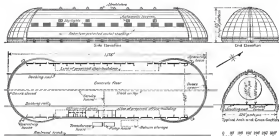
**A**S WORK PROGRESSES on the huge airship factory and dock being erected by the Goodyear-Zeppelin Corp. at the Akron (Ohio) Municipal Airport, it becomes more and more apparent that the structure is what its builders claim to be—the most unusual and interesting building under construction anywhere in the world.

At the time of writing, the dock is approximately half finished. At an early date it should present a fairly complete appearance to the eyes of the spectator. Work of assembling the first huge airship inside the building is to start within a short time.

The ZRS-4 and ZRS-5, two super-Zeppelins ordered by the United States Navy, will be constructed in the dock. It is highly probable that other air liners—commercial ships—will be built there also. It is also likely that, eventually, another dock displacing the present one will be erected beside it. At least, the company is anticipating such expansion by providing ground space.

The two new Navy airships will be the largest ever built. They will be 785 ft. long and 155 ft. wide, and will hold 6,500,000 cu ft. of helium gas. The dock is large enough to house a ship of 10,000,000 cu ft. capacity.

If the new building were sheet like a loaf of bread,



Plan drawing of the Goodyear-Zeppelin dock at Akron, Ohio.

the surface of each cross-section would be bounded by a parabola intersected by a straight line. If the docks were longitudinal and vertical, the section would be bounded by two half-parabolas connected by straight lines. The overall shape is much like a silencer's conical open longhouse. The building is of huge proportions. Its length is 1,175 ft., measured between center lines of dock tracks; its width is 325 ft., center to center of arch piers, and the height from the floor to the platform at the top is 251 ft. Area of the floor is 364,000 sq. ft., and cubic content of the building is about 34,000,000 cu ft.

The Zeppelin shed built in 1906-9 at Friedrichshafen, Germany, is 624 ft. long, 150 ft. wide and 62 ft. high. A later hanger built nearby is 787 ft. long, 150 ft. wide and 114 ft. high. These, like most other European hangers, are of structural steel arches with sliding doors at the ends. At Orly, France, are two somewhat different types of sheds. They are of reinforced concrete, and are 904 ft. long, 256 ft. wide and 294 ft. high. The largest British hanger is at Karachi, India, and was built for the proposed England-India airship service. It is 830 ft. long, 230 ft. wide, and 170 ft. clear height. The largest hangers in the United States other than the one at Akron are at Bolling, D. C., and Lakehurst, N. J. The Bolling hanger is 803x150x150 ft. That at Lakehurst is 803x163x172 ft. Both of the hangers have vertical, sliding doors opening from the center outward.

**I**N RESUME the Akron building, Dr. Karl Arnstein, vice-president and chief engineer of the Goodyear-Zeppelin Corp., had some precedent upon which to work. Similar structures, although on a considerably smaller scale, had been built and used at Dresden and Leipzig, Germany. A model in the general shape of the Akron building was constructed and put through rigorous wind-tunnel tests at the Dussel Guggenheim School of Aero-

nautics of New York University. The model was 1/250 the size of the completed dock. Tests were conducted by Dr. Arnstein and Dr. Wolfgang Klingsporn, the latter also of the Goodyear-Zeppelin Corp. Working with the knowledge that wind, in striking a building and being deflected upward, creates a suction that tends to force the roof of the structure upward, they found that the dock would have to be braced against tremendous force acting from the inside. Suction forces of this type sometimes are several times the magnitude of the wind pressure on the outside. Dr. Arnstein's experiments indicated that the outward force should be ignored at not less than one-half the inward, and that the roof should be secured locally against suction forces equal to the maximum direct pressure. This practice of designing a building to withstand pressure from within has not been applied extensively in America, suction being almost wholly neglected in structures erected prior to the Akron dock.

**M**EASUREMENTS made at the Lakehurst hanger revealed that it is unsafe to dock an airship when a cross-hanger wind is blowing more than 10 m.p.h. This velocity creates about the open, sliding-type doors, currents of over the stated wind velocity, or 20 m.p.h. Therefore it is evident that, if the door curtains can be brought to the same velocity as the cross wind, a 20-mile blow can be tolerated, or an effective increase in usefulness of the dock by 100 per cent obtained.

It was this factor which led to adoption of the "airtight peep" type doors for the new dock. They are designed so that a minimum of troublesome air currents is created, regardless of the door position.

Early in the airship's development it was learned that the hanger should be placed so that its longitudinal axis is parallel to the prevailing wind blast. The Akron building is oriented to north 30° east by north 30° west, approximately.

Calculation of loads and stresses for the building was in itself a huge task. The values finally arrived at include the following:

Dead loads—On arches, 30 lb. per sq ft. On sheets, rafters and purlins, 20 lb. per sq ft. On lateral, 20 lb. per sq ft.

Snow loads—On all surfaces having an inclination of less than 45 per cent to horizontal, 30 lb. in combination with dead load only, and 15 lb. in combination with maximum load.

Wind loads—On all sheets, purlins and rafters, plus 80 lb. per sq ft. for upper portion of building, and 40 lb. for lower. On all bracing and arches, a wind pressure of 37.5 lb. per sq ft. combined with internal forces of plus 12.5 lb. is allowed, and pressures are distributed according to pressure diagrams worked out from Dr. Aronson's experiments. A direct horizontal wind pressure of 15 lb. per sq ft. on external surfaces only, with no consideration for internal forces, is to be used only when stresses due to this load exceed those from wind load.

Working stresses for dead and snow load from beam to beam, 18,000 lb. per sq in. on structural steel, and 24,000 lb. on aluminum. For maximum stresses due to wind load, 24,000 lb. on structural steel and 32,000 lb. on aluminum, are the values.

Carbon steel used in the building has an ultimate strength of 55,000 lb. per sq in. and a yield point of 30,000 lb. per sq in. Aluminum used has an ultimate strength of 30,000 lb. per sq in. and a yield point of 45,000 lb. per sq in.

General design of the building finally was selected because of its aerodynamic qualities and because it is economical in steel weight. This design, as developed by Wilbur Watson & Associates, consists of 11 parabolic arches spaced 80 ft. on center, and connected by a system of vertical and horizontal trusses placed between the upper and lower chords of the arches, rather than in the plane of the arch chords.

These trusses, in addition to forming the substructure of bracing for the center shell of the roof, carry light trussed rafters spaced 10 ft. on center, and on these are placed 2-lb. purlins 6 ft. in center. At each end of the main shell portions are two diagonal arches, making the end arch span which are 800 ft. apart. The orange-pet doors are built up in a similar way, of arches and braced ribs. All materials are of structural steel, except chords of the main arches which are reinforced with aluminum steel.

**T**wo or the outstanding features of the design are the novel type of doors used and the absence of expansion joints in the building. Of course, such a large all-steel structure will undergo a considerable change in overall size with fluctuation in temperature. Therefore,

the main arches are fixed in position, while the remaining ones are carried on rollers. This permits the building to "stretch" and contract, much like a giant animal after a nap. The end arches, supporting the upper door bays, will move laterally about 4 in. under the maximum range of temperature. This motion is absorbed entirely on the joints of the bays and partly by the door structure.

**I**F YOU CAN CONCEIVE ONE-SIGHTS of an orange-pink shell stands 252 ft. high and is 214 ft. wide at the bottom, and is held by a large hollow forged pin at the top, you have a glimpse of the appearance of one door section. The curved base rests on 40 double-flanged wheels rolling on curved concrete set flush with the ground surface and resting on a concrete foundation supported by piles driven to bed rock. Weight of each section is 600 tons, or 1,200 tons for each end of the structure.

The supporting wheels are each of forged steel, 27 in. in diameter and the rolls upon which they rest weigh 100 lb. per yard. The rolls are set in standard railroad gauge. To avoid slipping of wheels with consequent wear the wheels are mounted individually, although positioned radially in pairs. Thus the vehicle wheels, having to travel farther than the inner ones, can make more revolutions without interference from the others, as would be the case if rubber or steel wheels were used, with solid axles. The wheel bearings house set against heavy springs so that, if one wheel receives more weight than its neighbor, the spring will give and tend to even the pressure. In such a large mass of steel, it is almost impossible to obtain perfectly even distribution of weight, engineers have found.

Door sections revolve about large pins at the top of the shell. These pins are spaced 4 ft. apart, and are raised to the roof girders by heavy steel frames. Each of the door pins carries no vertical weight, but serves



At view showing the completed ground operations building, greater capacity of the new dock.



An artist's conception of what the ground operations design and delivery on doors will look like when completed (looking northwest).

only to prevent the door from tipping inward or outward. Engineers have calculated that pressure against the pin will reach a maximum of 550,000 lb. inward, and 450,000 lb. outward, under the influence of snow and wind. Each of the pins is 12 in. in diameter and about 6 ft. long. The door section is fastened to it through a ball-and-socket joint, the ball member being 30 in.

An interesting feature of the pin is that its position is not fixed. It swivels about several inches in the building expands and contracts under temperature changes. The bearing also moves 4 in. up and down on the pin each way, to provide for thermal changes in the door height.

Moving such huge masses of metal as the door sections is no little task. In opening the doors, considerable power is required to start the sections in motion. Then, after they are moving, complete control must be possible at all times. Furthermore, the doors must not be permitted to move under wind action.

These conditions led to adoption of the "rack drive" system of door operation. Along the outside edge of each section, at the base, is a curved 14-in. H-beam, the radius of curvature being 152 ft. 10 in. The 4-in. in diameter are set along the beam at 9 in. pitch. The pins are engaged by the "half" wheel of the driving mechanism.

The four driving motors are located in concrete bases outside the door sections. They consist of a train of gears which operate a very large ball gear that engages in the door rack. The gear train is driven, through a worm gear reduction, by an electric motor. The worm gears are naturally self-locking, so that the door is held rigidly in a given position after the motor is stopped.

Each of the four door systems is built to operate at two speeds with a constant torque effort. It can drive the door at a speed of 25 or 40 ft. per min. The motors are General Electric, 200/100 hp., 15 min. rating, 600/300 r.p.m., 440-volt, three-phase, 60-cycle. A hydro-electric brake on the drive shaft serves to stop the motor, and is set to operate with a time lag of a few seconds after power is shut off, to permit the pins to slow down.

As operators desiring to open or close the doors merely press the proper button. Action from that point on is automatic. Electric control switches set on the runway area to the driving unit are operated by projections from the door. As the two leaves come together or approach the full open position, the switches are operated, and

the doors slow to a creeping speed, and ultimately stop in the proper position. This automatic feature makes possible the relatively high operating speed. In other hangars in this country, speeds of but 12 to 15 ft. per min. are used.

Roller bearings are not used on the door wheels because it would then be too difficult to arrest motion, once it is started. It is true that such bearings would permit easier starting, but the engineers decided to employ a greater starting power rather than to complicate the problem of control in slowing down and stopping the sections. It is estimated that in such an 80-ton pressure will be required to start the 600-ton doors in motion when snow and wind conditions are extreme, with the friction bearings used. Roller bearings also would not permit as high linear speed, because of the longer time necessary to stop the door motion. No trouble is anticipated from snow and ice getting into the rack and gear arrangement. The great pressure between the gear teeth and rack pins readily disperses such formations. The time required for opening or closing a door is estimated at 5 min.

**T**HE BUILDING is fundamentally a factory, and mechanical equipment is arranged accordingly. Handling devices are numerous, and include a 5-ton electric hoist running on a 12-in. I-beam at the center of the building, overhead. On either side of the center are two smaller beams carrying working platforms which can be adjusted to any desired height. Workmen will use these in assembling the airships. Other working platforms are carried on I-beams further down the curved building ends.

A fixed working platform extends from one end to the other of the building, in the center. On each side are six catwalks, each 2 ft. wide, running the entire building length. These are located between the chords of the arches. Access to the upper platform is by two stairways, one on each side near the center. There is also a specially designed inclined railway for men and materials. Two counterbalanced cars run on a curved track laid in such a manner that each car is horizontal at all times. The cars can be operated from the top of the railway or from either car.

The building width between arch pins is 325 ft., but only 240 ft. of this space will be required for assembling one of the Zeppelins. The remaining area of about 42x240 ft. on each side will be taken up by shop and office equipment. At first shops will be confined to the



see allowing, in each category, twenty points for the fastest competitor, ten points for the second fastest and five points for the third fastest.

The aforementioned table allowed no points for a speed under 52.8 m.p.h. and 50 points for a speed of 80 m.p.h. or above, with an intermediate scale for the first category, to require the second category, the figures were no points for 43.5 m.p.h. up to 50 points for 79.6 m.p.h. or above. In addition, competitors were disqualified for a speed less than 46.6 m.p.h. in the first category, or less than 57.4 m.p.h. in the second.

The final placings grouped the two categories according to the total number of points obtained.

Although somewhat intricate as it was, this formula worked satisfactorily at least inasmuch as it was that it was the first time that light planes belonging to two different categories were to compete together and it was accordingly difficult to place perfect rules at the first attempt.

#### Nationality of Competitors and Planes

THE INTERNATIONAL NATIONAL ASSOCIATIONS taking part were the Aero-Club de Deutschland (Prussia, Germany, Italy, Romania, Switzerland), A. volante allemande (Great Britain). But it was possible for a pilot whose National Association was not taking part in the Challenge to enter under the colors of a foreign Association of which he was personally a member, so British citizens entered under the flag of the "Aéro-Club de France" and others competed under German colors. In all, 62 citizens were fielded: four from Czechoslovakia, 2 from France, 34 from Germany, 14 from Italy, and 20 from Switzerland.

Makes of planes and engines may be summarized as follows: Czechoslovakia, two makers; with Walter 80 hp engines; Belgium, one maker, with Walter 85 hp engines; France, eight makers with Michel, Anson, Renault, Hispano-Suiza and Gnome engines; Germany, twelve makers with Siemens, Argus, Gipsy, Salomon, Gies, Cirrus, Wright Gale and Walter engines; Great Britain, two makers, with Gipsy and Cirrus engines; Italy,

five makers with Cirrus, Itala-Brambilla and Fiat engines; Japan, two makers with Cirrus and Salomon engines.

#### Preliminary Trials

THE COMPETITION had to be at Orléans Airport on Aug. 3, before news ready for the opening of the competition. 55 of the 82 machines entered turned out in time, and the weigh-in procedure started. All ships complied with the maximum weight limit.

The machines were next inspected by the judges for the sound of the "technical qualification" points. Out of a maximum possible total of 20, the last race was made by a Roman 5, with 85 hp Fiat engine, receiving 18.25 points.

On Aug. 5, the competition test took place in the town City-Airport-Orléans, to be covered twice one-stop (3027 m.). The 55 competitors were dispatched by groups of four, the fastest machines first, and were sent scattered along the course. Seven were compelled to enter owing to various mishaps, leaving 48 machines in the running qualified to undertake the four mile.

Performances varied to a great extent in the competition test. The best figure recorded was Nierberg's (Dornier D 18, 65 hp engine), 18.5 points out of 20 possible, at a speed of 57.5 m.p.h., next came Wirth (Klemm 25, 40 hp Salomon), 17.5 points, speed 49.1 m.p.h., Heister (Junkers A, 50, 85 hp Gipsy), speed 48.8 m.p.h., and Heister (Junkers A, 50, 85 hp Walter), speed 38.5 m.p.h., both with 17 points.

The preliminary trials were now over, before the start of the four, the provisional positions were as follows: the total number of points possible being 46:

1. Kleps (Avia 11, 35 hp Walter) ..... 32.5
- Wirth (Klemm 25, 40 hp Salomon) ..... 17.5
- Nierberg (Dornier D 18, 65 hp Gipsy) ..... 18.5
4. Lauer (Klemm 25, 40 hp Salomon) ..... 18.5
5. Heister (Klemm 25, 40 hp Salomon) ..... 17.5
- Rosler (Junkers 50, 85 hp Gipsy) ..... 17.5

On Aug. 7, the machines were based on the field and started by groups of four at 3.30 a.m. on their way to Biele. Weather was bad and pilots encountered blinding rain and windstorms.

It would be too long and tedious to attempt recording the position of leading cars all along the circuit. The progress of various competitors soon became very uncertain since every one was obliged to observe at the distance travelled each day.

At the conclusion of the preliminary trials, Kleps, in an Avia 11 with 85 hp Walter engine, and Wirth, in a Klemm 25 with 40 hp Salomon engine, each had 32.5 of a possible 46 points. Four other competitors had more than 25 points.

On Aug. 7, the machines were based up on the field and started in groups of four at 3.30 a.m. on their way to Biele. Weather was bad and blinding rain and windstorms were encountered. There was naturally a wide variation in speeds and positions along the way and it was impossible for men to tell who was leading at any point. But on Aug. 15, when the final check officially opened at Orléans, some twenty planes were already circling over the airport waiting for the signal, and as the motor was fired, there was a slow descent to the line in mass formation. A total of 25 planes arrived on that day before dark.

One machine came in on Aug. 15, five Aug. 16 and



A Dornier D 18, which made the fastest time in the four, 18.5 points out of 20, for the competition test. In the air it is a Dornier with which made the one performance for competition.

on Aug. 17, bringing the total number of survivors to 32, though one of these was subsequently disqualified for having failed to cover a leg each day. In all, therefore, 31 planes figured in the final trial.

The last average speeds in the first category were made by Carberry (Rush-Katzenstein 25, 85 hp Cirrus) and Caplan Broad (Moht, 35 hp Gipsy) with 90.4 and 95.5 m.p.h., respectively. In the second category, Moht (Rush-Katzenstein 25, 85 hp Cirrus) averaged 80.2 m.p.h. and V. Daugera (same plane and engine) averaged 78.4 m.p.h.

These four planes also had the best scores for the four, based on speed, reliability and number of stops, Carberry and Moht had 119 points each while Caplan Broad had 109 and V. Daugera 106.5 points. The final rating of the contestants is not yet absolutely determined, since the Italian government held a protest against certain competitors who were suspected of having flown over prohibited areas. It is about certain, however, that the following standing will be accepted:

[These results have been homologated since the article was written.—Ed.]

- |   |       |
|---|-------|
| 1. Moht (Rush-Katzenstein, R.P.C. Moht, 35 hp Cirrus) ..... | 129.5 |
| 2. Caplan Broad (Moht, 35 hp Gipsy) .....                   | 126.5 |
| 3. Carberry (Rush-Katzenstein, 25, 85 hp Cirrus) .....      | 119.0 |
| 4. Lauer (Klemm 25, 40 hp Salomon) .....                    | 118.5 |
| 5. Heister (Junkers 50, 85 hp Gipsy) .....                  | 117.5 |
| 6. Wirth (Klemm 25, 40 hp Salomon) .....                    | 117.0 |
| 7. V. Daugera (Rush-Katzenstein, 25, 85 hp Cirrus) .....    | 116.5 |
| 8. Moht (Rush-Katzenstein, 25, 85 hp Cirrus) .....          | 116.0 |
| 9. Caplan Broad (Moht, 35 hp Gipsy) .....                   | 115.5 |
| 10. Heister (Junkers 50, 85 hp Gipsy) .....                 | 115.0 |
| 11. Rosler (Junkers 50, 85 hp Gipsy) .....                  | 114.5 |
| 12. Nierberg (Dornier D 18, 65 hp Gipsy) .....              | 114.0 |

The first "Challenge de Tourisme International" was thus won by Germany, and the next one will be conducted by the National Association of that country.

#### The Trend of Design

THE MACHINES entered this year divided themselves in two very different classes: the "touring planes," with a considerable class cabin mandated for practical use in private owners rather than for competition, and the "sport planes," where comfort was sacrificed to performance. The latter was of course, although Capt Broad's plane which performed so well was a "Cope Moht" belonging rather to the first class.

Generally speaking, British, French and Italian ships were "touring planes," most of them with closed cabins, most of the German entries presented such characteristics and some were purely racing machines as the Rush-Katzenstein Dornier D 18, Bussard and Ando-Warwick. These last two were unable to take part in the

trial contest, the first arriving too late and the second breaking on the air whilst performing aerobatics. The Dornier was compelled to land during the Tour and sustained damage, while the crew was injured. Gies Rush-Katzenstein only completed the trial.

It was a disappointment not to see the two Cirrus Aviatours, their performance would have been watched with interest. None of them could be put ready to race. It was significant to note that machines of the second category (less than 600 hp engines) were much less numerous: only 18 out of 82 entries, and 8 out of 31 finishers. Such machines appear to be more suited for peripatetic and short journeys than for real touring. In the hands of exceptionally good pilots, they are able to hold their own even in worse conditions, but their use for private owners is somewhat restricted. Except in very special cases, the weight limit does not allow of an engine of more than about 40 hp to be installed; this appears too low to enable two people and a sufficient amount of luggage to be carried at a respectable average through any weather. The cockpit is open, which does not provide enough comfort. On this score, the appearance of passengers differed largely according to the type of cockpit used: those travelling in open places had to be clad accordingly, were often oil-stained and showed signs in some instances of marked fatigue. On the other hand, the closed type of cockpit was much more comfortable and remarkably fresh. Such facts mean a great deal for popularizing aviation.

In the large class (less than 500 hp engines), the power pool varied between 70 and 95 hp, light tests mean more rigid and comfortable, handles being apparently better. Some of the cabins were extremely roomy and well designed, allowing for side by side seats to be installed and leaving large space for luggage and spares. In some instances a third passenger could be carried for short trips, or additional fuel tank allowing of a very respectable range of 500-600.

#### Technical Results

THE MONOPLANE scored heavily against the biplane: 58 out of 82 entries (70.7 per cent) and 28 out of 31 finishers (90.4 per cent). Of these 28, 15 (53.6 per cent) were pure contenders, the remaining 13 being of the multiply-branded type. As against wing designers 15 (53.6 per cent) were low-wing, the rest belonging to the "parasol" type.

The construction was in the majority of cases wood and fabric with a few fuselages of welded steel. Some machines had timber covering throughout. There was one instance each of real "all-metal" construction, the "Junkers" which were of duralumin throughout, according to the well-known practice of the firm. Despite that unique feature the planes were not as heavy as might be supposed, with 85 hp engines, the weight did not exceed 1,225 lb, except for very reasonable achievement. But it is doubtful if this form of construction presents marked superiority, as the use of a light plane Duralumin is certainly hampered, but repairs are much more difficult as skilled metal workers trained in the handling of duralumin are to be found very rarely. On the contrary, a machine of wood and fabric throughout is usually so robust that, in the case of fuselage, both exterior and interior can be fixed easily, expeditiously.

It must be remarked that the folding or dismantling test was not compulsory, in fact, a certain number of competitors deliberately ignored it, thereby losing some



A "Fiat" being refueled through the "Aero" during the final test. Many more were being a fuel.

points, for the obvious reason that they were unable to comply with requirements of the rules in this case. On the other hand, others put up remarkable performances. It was necessary to fold the wings, wheel the plane under a mock-up "door" measuring 98.8 in. wide by 115.5 in. high and stretch the wings in less than 15 min. For each operation, the pilot and passenger only being allowed to undertake the task. The "Moth" and "Porter" did particularly well in this respect. In the opinion of the winner, the folding test should be made compulsory, a machine which requires complete dismantling to be wheeled through a gate is not a genuine touring plane and, sooner or later, its owner is certain to be in trouble on this score, probably in an adverse manner.

The undercarriage, most of them of the split-disk type, appeared in frequent cases to be made to cope with the stresses laid upon them by frequent landings on uneven rough ground. Too many heavy repairs had to be undertaken during the Tour of Europe; the welded steel construction proved here a special advantage over wood in distress. Shock-absorber systems were generally of the elastic cord type, even on the unusual "jockey" wheels where they are perfectly protected by dampers sliding bearings. Some planes used compressed rubber disks. Wheels and tires of too small dimensions were unsuitable. Very few wheel hubs were used, and in certain instances, they appeared rather flimsy.

#### Engine Lost at the Competing Places

Nearly a hundred water-cooled engine sets to be seen as might have been predicted. Of the 33 fighters 25 (80 per cent) were three cylinders and 6 (19.4 per cent) four-cylinder. Of the 25 fighters 12 (52.8 per cent) had five cylinders, 12 (38.7 per cent) had seven cylinders and 5 (16.1 per cent) had nine cylinders. It must be remarked that the smallest engine represented, the Deth 40 hp. Siemens, had nine cylinders.

All the propellers were of the wooden type, and direct drives, although some engines were turning at somewhat high speeds: 2,000 revolutions and more were not infrequent. Taking in consideration the moderate speed of the machines (under 100 m.p.h. in nearly every case), it would appear that a possible increase of efficiency could be secured by using reduction gears.

Too many engines still require control of fuel through lubrication, for touring planes, the use of ordinary material oil of the type used for cars is practically a necessity.

Many machines had no starter and relied solely on

propeller swinging by the passenger (who was always a skilled mechanic). This method cannot be accepted for a touring plane worthy of the name and it is desirable to have regulations compelling attempts to use electrically actuated starters operating from the seat, without touching the propeller in any way.

The efficiency of these small engines proved wonderful. After observation of all those which were satisfactorily prepared, a procedure which took very short time, the remainder turned out splendidly, despite the severe grinding imposed upon them. On some machines, they were called upon to deliver nearly full power most of the time, and they did it unflinchingly. Not only the engines proper, but all the accessory components of the power-plant functioned excellently, irrespective of make and type. It is significant to note that engines were seldom broken beginning at the contest and had to remain so until the finishing line was passed; only one competitor out of 31 failed to show airplanes with at the end. Such reliability from a number of different types of engines was well expected, even by the most optimistically minded, and the fact gave evidence of the great strides made by multi-cylinder engines.

#### General Remarks

THE CONTEST was generally considered beforehand as a track too severe for light planes, many were of the opinion that a circuit of 3,000 mi., embracing some very difficult fields, would cause dangers amongst the ranks of competitors. They predicted that the number of survivors would be very small, a fact not likely to enhance the position of the light plane. It was optimistically estimated that the fighters would not exceed the dozen, putting things at their best.

Circumstances, however, proved considerably such views to be considerably on the pessimistic side and the promoters are to be congratulated in having given modern light planes a complete opportunity to demonstrate their possibilities in a most severe contest. It must be admitted that nearly all the pilots were old hands, although among them was a girl, Miss Spooner, and some genuine private owners. But the average structure is not likely to expect of his machine and branch one-fifth of the best test with which planes and pilots were faced in the Challenge. Signs which coupled with flying orders from the trial are good and reliable machines capable of giving reliable service in the hands of a normal customer, and such a demonstration has its value.

Present light planes have their shortcomings which are no longer perfected. But proof was given that a

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small two-seater with ample accommodations and reasonable cruising speed does not require more than 80 hp. and therefore is not so expensive to run. Some figures of the competition have been introduced to read wide as the subject. The star performer was not at all, as the "Dunstable," which achieved it, is a beautifully designed performance job, but in no way a practical touring plane and few machines would be able to average 92.4 m.p.h. on a closed circuit 203 mi. in length with a consumption of only 61.5 lb. of fuel. Apart from this exceptionally good result, the averages were as follows:

First category, speed about 82 m.p.h., 85 lb. of fuel. Second category, speed about 71 m.p.h., 54 lb. of fuel. These figures represent roughly 42 and 26 lb. of fuel for 100 mi. at a fair speed, sufficient for touring purposes.

Incidentally, the average figures recorded were: First category, best 61.5 lb. at 92.4 m.p.h.; worst, 112 lb. at 92.4 m.p.h. Second category, best 46.4 lb. at 69 m.p.h.; worst, 72.9 lb. at 85.4 m.p.h.

During the 1928, some patches of nearly atrocious weather were encountered, especially at the beginning, but the little machines quickly forced their way through, even the lowest ground. Some forced landings resulted, in most cases without damage. It will be remembered that points were granted for pilots who were selected as non-competitor, contrary to expectations, only one of the 31 fighters failed to obtain maximum score on this.

Accidents during the actual contest were remarkably few and not one was serious. As already mentioned, a machine broke in the air while straining before the comparison flight, and the pilot, Herr Hoffmann, was unfortunately killed. He had not thought 61 to take with him his parachute, which would probably have saved his life. While on the subject of parachutes, it is interesting to note that the crew of certain machines, rather of the racing type, considered them as normal landing means in case of engine failure while above difficult country. Surely a "banned" frame of mind!

Some of the scheduled landing fields were not too big, but take-off apparently went well even there, although certain machines were generally loaded. It must be mentioned that the "Auto-Club de France" asked judges the opportunity for publishing the first "Guide for Air Tourists," a very useful booklet containing detailed information about all the airports on the circuit, with detailed plans of them, heights of obstructions, and a map showing exact position of the field in respect of the nearest towns.

#### The Future of the Challenge

THE CHALLENGE is planned as a yearly affair. But the regulations are not unchangeable and it is certain that this year's rules will be altered for 1930. The main difficulty is the question of the two categories. It would appear more logical to make two different classifications, but the contest would lose much of its value if there were to be two winners. This summer, the rules tried to place the two classes on an equal footing by means of the different scales of yards for consumption and speed, but this good intention was practically negated by a fact which rendered useless the provisions taken: some of the lighter machines—the ultimate winner amongst them, of course—were equipped with engines of nearly the same power as the heavier planes. It is all right for the designers who succeeded in keeping their structure weights as low as is permitted of the use of metals rarely



Just after the Dunstable Air begins the test. This machine was the only aircraft to place in the contest.

despite the power of those used by less fortunate rivals, but, from a general point of view, the fact completely spoils the balance between the two classes. Something better ought to be found for next year.

As regards importance of speed, some criticism was cast upon the 70 points (out of a maximum of 165) granted on this score. In fact, the deciding factor for final places was the distribution of twenty, ten and five points to the three fastest planes on the tour in each of the two classes. It seems likely that these supplementary points for fastest times will not be used next summer.

Finally, as regards the planning for awarding the Challenge Cup, the present regulations are not free of criticism. The cup goes to the National Association to which the winner belongs. It would appear fairer to establish a real time classification, by providing for example that the three first pilots from each National team count for the Challenge Cup. In this case, a limited number of pilots should be officially recognized by each National Association before closing of entries, as it is the case in many other sporting competitions. Incidentally, the result this year would have been: First, Germany: 1, 3, 4, 8 points; second, Italy: 5, 8, 9, 22 points; third, France: 2, 13, 22, 34 points.

Saunders up the first "Challenge International des Touristes" was the biggest and one of the most interesting aviation competitions ever witnessed in Europe, it proved conclusively the status of modern light planes and engines. Many splendid lessons were learned from a by-players and its future appears very promising.

The experience gained this year will enable an improved set of rules to be drawn for next contest. One must express the wish that the entry list be of an even more international character: some European countries were not represented adequately this year and the fact is to be regretted.

Still more, no representative of America was present. It is to be hoped that next year some models of American light planes will enter a competition which could then boast of being draped by the best light planes of all the continents of the world!



The 48 competing planes lining up for the start of the Challenge des Touristes International, one of the largest and most interesting aeronautical events ever witnessed in France.











## BRIEFLY

Proposals for creating a 200-ft. monument to be dedicated at the top of the Empire State Building which is to occupy the site of the Waldorf-Astoria in New York City are being considered by the board of the Navy Department.

Construction of four additional aircraft carriers for the Navy has been recommended by Secretary Adams. Hearings on the progress of the Enterprise Army aviation program will be held beginning Jan. 7 by the House of Representatives Committee on Military Affairs.

The U.S.S. Lexington has arrived in Tacoma where it will stay until it helps to supply the city with electricity, due to the failure of the city's hydro-electric system because of drought.

An experimental rate for a \$10,000 cap donated by Thompson Products, Inc., will be held hereafter in connection with the National Air Races, according to a report of a recent meeting of the N.A.A. committee. Chicago has been tentatively selected as the site of the races in 1963.

A new athletic landing gear for Parks Park flying plane has been successfully tested and approved by the Department of Commerce.

Only three per cent of passengers on T.A.T.-Vladivostok have suffered from air sickness in November, as compared with seven per cent last July.

Plans for a non-stop relative flight from New York to Buenos Aires have been announced by Herbert H. Pomeroy. The attempt is to be made soon.

Daily newspaper service between Seattle and Yakima, Wash., will be resumed by Inter City Air Express Corporation.

Five at the Tampa, Fla., municipal airport are scheduled in a two-extended at \$10,000.

An automobile device to enable pilots to pick up mail while in flight, invented by J. J. Jacobs, will be tested by Southern Air Transport at Fort Worth, Tex.

Western Air Express is offering a reduced 20 per cent discount between Kansas City and Los Angeles during the Christmas holidays.

Reports from Washington indicate that the U. S. State Department will require to leave Blue Island to the city of Philadelphia, at a nominal rental for airport purposes.

A new office address of Detroit Aircraft Corporation is to be established at Farmington, Conn., according to a recent announcement by Edward S. Tenna, president.

Penfold George & Bob Company has begun production of thick aluminum fuel nozzles, but further production will not be started until next year.

Canadian Airways, Ltd., has inaugurated daily air mail service between Montreal and the Hawaiian Islands. Contract for carrying mail between

Ottawa and Chicago has been awarded to Compuair Aviation Finance Corporation.

A recent arrival of jets at Love Field, Dallas, Tex., caused some excitement at El Paso.

An express service between Pittsburgh, Pa., and Washington, D. C., was inaugurated Dec. 16 by Colonial Rail, Inc., and Railway Express Agency.

Delaware Line Nevada and Idaho report Dec. 16 to discuss, among other projects, establishment of an airport between Las Vegas and Reno Falls.

Plans also have been made that the U. S. in Sidney, Washington, will receive gold bars from the company. Hearings will occur early next year.

Plans of Pacific Air Transport Company are now delivering shipments of high speed to carriers such as Alaska, Pacific, Canadian, and others, each having their own 25-in. jet engine.

The Aviation Corporation reports a net loss of \$335,389 for the quarter ended Sept. 30 after depreciation and amortization charges.

Consolidated Air Lines, Inc., will soon receive a new jet engine from Los Angeles to San Antonio and Houston, Tex. The company has made a 25 per cent reduction of fares during the holidays.

Weather has been interfering with transport operations in many parts of the country. Lines operating from Texas to Chicago, Richmond, and its various parts of Canada have reported several delays during the past two weeks.

First planes to be built in the Vancouver plant of Boeing Aircraft of Canada, are being completed and will be shipped to Los Angeles.

Airway aviation headquarters at Boston Airport has been supplied with a new code.

## AERONAUTICAL CALENDAR

|              |  |
|--------------|--|
| Jan. 10-11th | U.S. Air Force Meeting, Miami, Fla.    |
| Jan. 11-12   | U.S. Air Force Meeting, Detroit, Mich. |
| Jan. 14-15   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 16-17   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 18-19   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 20-21   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 22-23   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 24-25   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 26-27   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 28-29   | U.S. Air Force Meeting, New York, N.Y. |
| Jan. 30-31   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 1-2     | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 3-4     | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 5-6     | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 7-8     | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 9-10    | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 11-12   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 13-14   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 15-16   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 17-18   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 19-20   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 21-22   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 23-24   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 25-26   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 27-28   | U.S. Air Force Meeting, New York, N.Y. |
| Feb. 29-30   | U.S. Air Force Meeting, New York, N.Y. |
| Mar. 1-2     | U.S. Air Force Meeting, New York, N.Y. |
| Mar. 3-4     | U.S. Air Force Meeting, New York, N.Y. |
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| Mar. 25-26   | U.S. Air Force Meeting, New York, N.Y. |
| Mar. 27-28   | U.S. Air Force Meeting, New York, N.Y. |
| Mar. 29-30   | U.S. Air Force Meeting, New York, N.Y. |
| Mar. 31      | U.S. Air Force Meeting, New York, N.Y. |

outboard motorboat for use in case of forced landings on the water.

United States Airways, Inc., has been given a franchise to operate a passenger line between Wichita, Kan., and Oklahoma City, Okla., with land as an intermediate stop.

Miss Joan Ray Shedd recently completed a transcontinental auto flight, leaving Los Angeles, Calif., on Jan. 1, 1959, with Wright J-5 engine, in 30 hr. flying time.

Palmer Airport, Inc., Kansas City, Mo., recently declared a cash dividend of \$35 per share, payable March 1, 1960, representing profits from the sale of assets to the company.

Park Air Lines has purchased two planes from Gillette, Inc.

Experimental daily air mail service between Montreal and New York, N.Y., was scheduled to start Dec. 9.

The name of the Los Angeles municipal airport has been changed to "The Los Angeles Airport."

Varney Air Lines plans to establish transport service between Spokane, Wash., and Seattle, Wash., in connection with Western Air Express and Pacific Air Transport, since that line has been closed.

Recently have been made to have Utah (N.Y.) municipal airport designated as a port of entry into the United States.

Thompson Aircraft Corporation plans to equip all of its small planes with radio.

Transair Air Lines, San Antonio, Tex., report 30,000 in of flying by company planes during November.

Pioneer Aircraft Company, Omaha, Neb., is planning to establish a branch office at Tulsa, Okla., and Kansas City, Mo.

Calcraft Airways plans to fly 30,148 mi. during November, carrying 31,484 lb. of mail and 470 passengers.

The Post Office Department is considering making Rockford, Ill., a stop on the air mail route between Chicago and the Los Angeles.

On a recent 275-act test flight from Chester Field, Buffalo, N.Y., to Detroit, Mich., and back, the flight was a success, demonstrating that the plane is a success.

Plans for building a factory and manufacturing airplanes are being made by W. E. Blanton, Joe Manufacturing Company, New York, and others, at Princeton, N.J.

Western Canada Airways, Ltd. has been given a franchise to operate a passenger line between Vancouver, B.C., and Seattle, Wash., for use as the Winnipeg-Seattle service.

Florida headquarters for Suncoast Aircraft Corporation have been established at Daytona Beach, Fla., by Mr. J. Raymond, distributor, who also plans to open a branch of United Air Lines at Jacksonville, Fla.

Sales in the southeast have been decided by Pan American Petroleum Company as a result of use of a Sikorsky helicopter in the region, according to company officials.

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## PERSONNEL

STEWART MORRIS, recently general traffic manager of Colonial Airways, has been made aviation traffic manager of The Aviation Corporation. BEN A. POLLEY becomes traffic manager for Colonial Airways.

JOHN W. BLAIR, Jr., has been made vice-president of the Aviation Corporation. BEN A. POLLEY becomes traffic manager for Colonial Airways.

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## SCHOOLS AND COLLEGES

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## AIRPORTS AND AIRLINES

Southwest Mail  
Routes AnnouncedFast Airways Selected  
For Early Development

WASHINGTON (A. C.)—A decision on the southwestern transcontinental air mail route, on which a hearing was held Nov. 25, has been reached by the Interdepartmental Committee on Airways. As yet, however, no aviation, Dec. 14. Four airlines will be linked together to offer a complete service to the Southwest. The route from the West will pass along the border, going from Los Angeles through San Diego and El Paso to Dallas and Fort Worth. From Dallas it will proceed to Austin, Sherman, Jackson and Birmingham.

Audiot route will be established from Los Angeles through San Diego and Dallas to Los Angeles, and passing through Little Rock, Memphis, and Nashville. This line, which will link with the Louisville-Chicago route, will run via a route parallel to the Chicago-Fort Worth and the New York-Albany line, but at some distance from both of them.

Orleans will be connected with the main western route by a line running from San Antonio, to which the San Antonio route will be connected, to a point on the line between El Paso and Fort Worth. Meanwhile, will probably be secured as a winter route for this line, although Southwest and the Springs are also serving connections.

## A South Coast Line Planned

The last of the four lines decided upon will be published between New York and St. Louis, with stops at Philadelphia, Pittsburgh, Columbus, and Indianapolis.

An estimated cost for the setup of these four lines places the expense at approximately more than \$10,000,000. The Associated Airways officials believe that it will not be necessary to ask for a deficiency appropriation to cover the cost of establishment and maintenance. In this connection it may be noted that the budget for 1931, now pending before the House Appropriations Committee, carries an increase of \$1,238,380 over the appropriation of the present fiscal year, to be used for maintenance and operation of lighted airports to be established by June 30, 1930. The requisition of \$200,000 of new money in 1931, and their maintenance and operation for part of the year.

The New York-St. Louis route may be operated by the spring of 1930, it was indicated on part of this line

## Boeing On Location



Several sets of the Boeing model for further observation points, which serves full meteorological equipment for the study of conditions in various parts of the territory flown over by the company's planes.

has been set up already. An offer has been made by the state of Pennsylvania to secure landing fields throughout the state and place them at the disposal of the government at a nominal cost or none.

Survey work on the western route will probably be started this spring, in an effort to complete as much of the preliminary detail as possible before the appropriation becomes available on July 1, 1930. Approximately a year is necessary for the complete establishment of a route.

F. C. Henshaw, chief of the Airways Division, states that the work at equipping the route will probably start from mid coast and west on. At the present time it is contemplated that the route on the coast will be established by the end of the fiscal year. This will necessitate equipping both ends of the southwestern route for night flying. Should delays occur over the equipment to be used at the end of the route, it will have to be used by trial or built over with the next day. Eventually the whole route will be equipped for night flying, and a double schedule established. The route from San Antonio to the point between El Paso and Fort Worth is expected to be flown during the day. The next route to be established is expected to be flown by the Airways Division in 1930. This includes three of \$250 for lighting, \$200 for maintenance and \$250 for other expenses.

(Continued on page 1235)

Pacific Zeppelin Firm  
Elects New Directors

NEW YORK (A. C.)—Twelve additional directors have joined the board of the Pacific Zeppelin Transport Co., Ltd. They are as follows:

Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York; Mr. J. B. Dwyer, Director, National City Bank, New York.

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Air Mail Shows  
Drop in November

WASHINGTON (A. C.)—For mail decreased 78,664 lb. in November from the volume carried in October, according to the monthly report of the Postmaster General. The figure for November was 624,141 lb. as compared with 702,805 lb. for the previous month. It should be noted, however, that there were less 25 business days in November as compared with 27 in October. The figure for October was 20,772 lb. in November. The postmaster for September was 666,062 lb. in August. The preliminary monthly report just issued is as follows:

|                      | No.   | Sept  | Oct   | Nov   |
|----------------------|-------|-------|-------|-------|
| Boatmen's V.         | 1,000 | 1,000 | 1,000 | 1,000 |
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## FOREIGN ACTIVITIES

Free Crash Repair  
Planned by DeHavilland

LONDON (ENGLAND)—The DeHavilland Aircraft Company has announced that beginning Dec. 1 of this year it would undertake to repair free of charge its aircraft on all days to meet the needs of airlines from a forced landing caused by the failure of a DeHavilland Gipsy engine. If the engine and airplane were delivered new and unused in the previous or other Dec. 1, 1959. This annual contract follows the recent successful aircraft reliability test to which a 154th fixed-wing Gipsy engine was subjected. The engine, a six-cylinder, will be constructed, rate better than 500 hp, with practically no requirement for oil or other liquid in the usual routine operation.

There are certain objections to this proposal, of course. The repair contract will be in effect 12 mo following date of delivery of the machine and engine. It covers only British plane having a British certificate of airworthiness and registration and only machines being used for private flying, training, air work and military operations. Racing, race and sport flights are not covered.

The service does not apply to scheduled airlines unless the company at its discretion decides to transfer the agreement to new companies. The agreement upon the record of the machine and engine to be transferred. The agreement also covers only failures of engine parts which are manufactured under the control of the DeHavilland company directly. This would rule out radio engine, auxiliary engine, fuel, carburetor, gasoline and oil lines, etc. It is stipulated that maintenance is accorded with the DeHavilland standards and that the DeHavilland engine be the owner previous to the crash.

## Poland Plans Lighted Routes

WARSAW (POLAND)—Plans for lighting various airports in Poland have been the subject of study for some time by Engineers Kuz and Pruszyński, as pointed by the government. During the next five years, the government expects to spend \$2,000,000 on the lighting of various airports. According to the most recent announcement, the first line to be lighted will be Zamość-Poznań-Warsaw-Lodz, to the Borek, the first. Plans are also being made for a line from Poznań and Lodz to Berlin, Warsaw and Moscow, with a branch to Leipzig, Commerce, Bucharest, and by way of the Black Sea to India. The Polish part of this route, and eventually all of it, will be lighted, according to plans.

## Aviation Increases in Iceland

REYKJAVIK (ICELAND)—Signs to increase in capital to 200,000 crowns (about \$220,000) have been taken by the Icelandic Aviation Company. Five additional Junkers airplanes will be added to the two now in service, and flying activities will be greatly increased at the beginning of next season, according to the announcement.

French Developing  
Equipment of Airports

PARIS (FRANCE)—Equipment at Le Bourget (Airport) is becoming well suited to cope with ever increasing traffic. Part of the activity granted by the government for civil aviation is devoted to establish enlarged buildings and concrete sheds, as well as for extending areas of traffic platforms.

The Garenneville Airport, near Cherbourg, is to be much enlarged and the Cherbouge-Orville terminal will be modified so as to give more room for the buildings. This field is particularly important by reason of the trans-Atlantic passenger traffic. Special plane services are being established for local and foreign airports.

The Chamber of Commerce of Rouen has decided to allow a sum of \$5,000 for its participation to the construction of an airport near the town of La Madeleine.

## Tail-Less Plane Built by Lippich in Germany



DRIVER CONSIDERABLE at the controls of the machine, built by Hary Lippich, at the time of the recent test flights at Tempelhof Airfield, Berlin. Note that the ground control and the close resemblance to the glider type construction from which the machine was evolved. The radiator are located at the wing tips.

British Investigate  
Evaporative Cooling

LONDON (ENGLAND)—Considerable attention has been paid recently to the value of evaporative cooling in reducing the weight to power ratio of the water-cooled engine and improving efficiency generally. The British Aeronautical Research Committee has been carrying out tests with steam radiators, both on the bench and in the engine, using a standard engine. The committee reports that the engine plane radiators results as regards fuel economy and lubrication on bench trials at 35 hp and flight trials at 35 hp, using a normal lubricating oil of the kind of the usual treated engine oil.

Wind tunnel tests have been made on a steam radiating wing, radiator cooling on English Electric P-10, has decided to enter the aviation field. He is to concentrate on an area engine of medium power likely to command a ready world sale. So far no details of his plans have been made public, but it is learned upon good authority that he proposes to market a seven cylinder air-cooled radial of about 145 hp. It will be a ground engine, and a strong case in the induction system should impart a small supercharger effect.

WMI statements have appeared that he was going to bring new engine prices down to approximately \$5 per horsepower, but Sir William Morris is laboring under no such delusion. He has engaged technical consultants, has prepared the designs, but realizes that the manufacture of an engine cannot be compared to the mass production of a car engine. Still he expects to set the price by about half that of any existing engine of about the same power, and it is understood, is prepared to sell large sums in losing down a present trading plant to achieve the low cost essential to a comparatively priced engine intended for world sale. The first engine is expected in about six months time.

## Esperandou Tail-Less Plane Ready For Tests



WELL-KNOWN for his glider activities, Herr Esperandou has extended his interest to the tail-less type. He is shown here in his two-place machine at recent tests at Tempelhof Airfield. Note the wheel undercarriage.

English Auto Man  
To Build Aero Engines

LONDON (ENGLAND)—Sir William Morris, who may not yet widely be known as an English Henry Ford, has decided to enter the aviation field. He is to concentrate on an area engine of medium power likely to command a ready world sale. So far no details of his plans have been made public, but it is learned upon good authority that he proposes to market a seven cylinder air-cooled radial of about 145 hp. It will be a ground engine, and a strong case in the induction system should impart a small supercharger effect.

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## Mexico Expands Air Service

MEXICO CITY (MEXICO)—An announcement is made by the department of aeronautics of the Ministry of War and Marine that 4,000,000 pesos (about \$2,000,000) will be spent during the next year in the development of the Army Air Service, both for the purchase of new planes and for equipment for government aircraft.

This represents a considerable increase over the appropriation for last year, and it is estimated that by the middle of next summer the Air Service will be more than double its present size.

## Record Made to Mexico City

MEXICO CITY (MEXICO)—Pilot Carlos M. Drayton of the Mexican Aviation Company bettered his own record for the trip to this city from Brownsville, Tex., when he landed a heavily loaded Fieseler here last week Jan. 16, after about leaving the Texas coast in a 15 hr. 35 min. flight. The air distance between the two cities is 450 mi. The pilot made previously good time on the 525-mi run between here and Tampico, negotiating this mountainous terrain in 2 hr. 5 min. The plane carried 16 passengers, a cabin operator and steward, 155 lb. of baggage, 25 lb. of cargo and 71 lb. of mail.

Announce Tempelhof  
Post Operation Figures

BERLIN (GERMANY)—Operations at the Tempelhof Airport during the business year 1958 resulted in a net profit to the airport holders (City of Berlin, Federal Government and State of Berlin) amounting to 382,825 marks (\$41,125), according to a report obtained by the U. S. Department of Commerce representative here. Of this total profit, 191,336 marks (\$20,385) went into starting and landing fees, 149,100 marks (\$15,860) from rental of hangars, repair shops, and other services, mainly in Deutsche Luft Hanse. Both of these items represented an increase of 100,000 marks over the year 1957. Starting and landing fees rose 7.5 marks (41,725) and 9.6 marks (10,340) for single and double landing places.

Construction work during the year cost 1,563,000 marks (\$159,600), and most notably in an enlargement of the administrative building to include longer restaurant space and hotel rooms. Especially it is expected to be five times its present size. Runway space in front of the hangars was enlarged, and the area of the field increased to 1,380,000 sq. m. (341,600,000 sq. ft.). Plans are now being made for placing new taxiways along all old structures in the vicinity.

Plan Air Service  
For French Indo-China

PARIS (FRANCE)—The Compagnie Air-Asie, headed by M. Jacques Huguier is preparing, in connection with the Compagnie Air-Indochine and the Compagnie Air Union-Lignes d'Orégon for the establishment of a new joint venture flying to Indo-China, as well as a set of airlines within the large French colony. The line is to pass through Calcutta, Agaña, Bangkok, Hong Kong, Yenchow, Hanoi, Haiphong, and will be extended later towards China and Japan, by Canton, Shanghai and Tokyo.

For the air in Indo-China, three routes are contemplated: 1. Pithouville, Bangkok, Angkor, Pnom-Penh, Saigon; 2. Hanoi, Yenchow, Tientsin, Singa, Hong Kong, Saigon; and 3. Lening-Peking, Hong Kong, Sumatra, Kuala Lumpur.

In addition, a very interesting arrangement of relieving traffic is to be opened next April. The arrangement from Hanoi to Indochina means three days in Saigon harbor, where shipping and aviation are tied in interesting things to visit. The magnificent sunset views of Angkor are but 200 mi. from the French border, and the best means of transportation are for two days to make the return journey to be made inside the available time. A plane service will be offered to passengers to visit the ruins and a day in Angkor-Thom and Angkor-Thom, and make a thorough visit of these places. Another air circuit for tourists is intended for travelers from Paris to visit at Bangkok, the first-class being again Angkor, 255 mi. away.

## Imperial Airways Cuts Prices

LONDON (ENGLAND)—Substantial reductions in the rates on Imperial Airways flights from London to the Far East are announced. The fare on the 8-m. airship is now only £15 (117s.), less than the £20 (160s.) which was the fare for the last season, leaving this city at noon daily, the fare has been reduced to £15 (117s.), less than the £20 (160s.) which was the fare for the last season, leaving this city at noon daily, the fare has been reduced from £20 (160s.) to £15 (117s.).

## French Line Improves Schedule

SAO PAULO (BRAZIL)—Daily passenger and mail service between Sao Paulo and Rio de Janeiro is planned by Compagnie Generale Aérienne, according to an announcement by the Brazilian Air Force. The Brazilian government intends to operate and mail connection has been made. It was reported that the Brazilian Air Force and the Brazilian Air Force would negotiate a 10-weekly service over the same route.

## P.A.A. May Extend Line

MIAMI (FLA.)—Extension of Pan American Airways' line from Miami to Rio de Janeiro, Montevideo and Buenos Aires is now under consideration, it is reported.











## SIDE SLIPS

AVIATION  
December 20, 1939

By  
Robert R. Osborn

THE DENNIS' ANTIPODE stopped by the deck just now for his weekly supply of cigars and in the course of his great sales talk showed us a shipping form from a local paper concerning the proposed flight of the Great Zepppelin to the North Pole next spring.

"Under command of Captain R. A. Lehmann, the Great Zepppelin with the twelve crewmen, will carry fresh provisions for five days—the longest period it is expected to remain away from its base at Tromsø, Norway, and Fairbanks, Alaska. Emergency provisions for ninety days will be carried and full equipment for travelling on foot on the ice. Twenty three extra days will go along."

The Starved Aviator says that before he went on that expedition he would want a clear understanding whether extra days were allowed in case provisions or emergency provisions.

Mrs. R. A. C. of Lyndhurst, N. J., sends in some clippings about the department store Santa Claus in Wilkes-Barre, Pa., who jumped with a parachute from an airplane over the town.

He intended to land in a very park where thousands of children were waiting for him, but missed badly and fell into the Susquehanna River, from which he was rescued by a policeman. It had always been our conclusion that Santa Claus was very inaccurate in his handling of his face-pack, and now it seems he is just as inaccurate in his work with the seat-pack.

Latet, N.A.G. of the Palo Alto School of Aviation Palo Alto, Cal., sends in the following copy of a letter of inquiry received by his school:

"Dear Sir—May I ask you whether your school will permit a man who wish to complete his Primary course, by doing some short work for your school, of course for some other course like flying which I am willing to pay which can not be rely upon you, which is very expensive. Training you will consider my application as is unimpaired. Very truly,

One who has become too "air minded."

ONE MARTIN FLYER

DEPARTMENT

Mr. H. L. C. of Garden City, N. Y., wandered into the office the other day with a great story about an abandoned pilot, which story was told to him by P. K. of St. Louis. The abandoned subject of this story is a Navy pilot but at the time of this incident had been perishing in grandeur all of his time in land planes. He was transferred to the San Diego station and was again assigned to landplanes for some time.

Then he was given a Martin T-42 seaplane, and was then chosen as head of flight started to set the ship down on the hard field. His co-pilot took the ship away from him, pulled it off and made a landing out at the ocean. The abandoned man was extremely apologetic, explaining that he had had to much time in landplanes that he was always forgetting himself when he got into a seaplane. Then he climbed over the side and dropped overboard.

An Oakland California man is making the distance on the grounds that his wife is "million, man, millionaire, married, disengaged, messy, grown-up, cool, better, golden, breeding, person, handsome, beautiful, brilliant, mostly, glorious, imperious, selfish, contemptuous, insensitive, snooty and inconsiderate."

This position of the vocabulary must have been the author of most of the comic descriptions along some new air line, and other provisions of some new airplanes we have read recently.

We are very much pleased with the progress made by the Society for the Aeronautical Education of Newspaper Correspondents and Artists. This society, of which we are all the president and only member, was started on its educational campaign by the featured and wonderful portraits of airplanes which made their appearances in the news columns and magazines after Commander Byrd's North Pole flight. Then he and Bennett were pictured flying to the Pole as everything between

a twin engined Jenny with our captain just forward of the tail to a Sea Gull flying boat fixed with trailing board dais.

The various news accounts were equally bad, reporting the Commander rambling along the outside of the ship to the tail to make useless observations, and Bennett top-entrancing an ordinary motor which had developed serious oil leaks. In the interests of the younger generation, which might grow up thinking that airplanes really look like those pictures, we took up the fight and devoted this valuable space for, in these many years, to the aeronautical education of newspaper reporters and artists. The work has been hard and the reporting. Many times just as we thought some progress was being made, some paper would print a picture of a ship spraying tail down, or a cigarette advertisement would show the gas hose end of a ship being refueled, showing conclusively on the top wing of his ship.

New the Commander had flown to another Pole, and with great fan and timing we watched the papers. The results were absolutely astounding. The only serious social error which we have found in far was called to our attention by E. A. H., is the San Angelo, Texas, Evening Standard, which shows a picture of a Parker Universal in the "top to-bottom plane" to be used in the flight. Another disappointing feature of the flight really can't be blamed on anyone but it is disappointing, nevertheless. On his North Pole flight Byrd saw "no life nor sun" according to the account New York Times. We have read many of the reports of this flight appearing in the Times but none of these clear up the life and sun situation at the North Pole. All of these years we have been waiting with great anxiety for this information, and the Times' choice to contribute that piece.

Well, we can't accomplish everything in one lifetime, and in general we consider that the work for which the society was organized has been satisfactorily accomplished. As soon as we can dispose of the office equipment, a large wing basket, the society will retire to function. Anyone willing to contribute to the fund, however, may make out personal checks to the editor at this again, and the money will be put to good use immediately.

AVIATION  
December 21, 1939

## The PARKS TRAINER



## STANDARD TRAINING PLANE AT AMERICA'S LARGEST AIR COLLEGE

Standard training plane at the Parks Air College—largest and finest in America—the Parks Trainer is now available to independent flying school operators and to all operators. Especially adapted to student instruction—ideally designed for private pilots—it is the foremost plane of its class on the market today.

Back by a division of the Detroit Aircraft Corporation, the Parks Trainer is backed by an organization which holds a position of acknowledged leadership in the industry. In design and performance it is recognized today as the most satisfactory type of training and sport plane.

Such commanding features as inherent stability and amazing maneuverability—a startlingly low landing speed and 50 m. p. h. cruising speed—combined with positive flight control are a few of the many superlatives of the Parks Trainer.

The Parks Trainer—equipped with an OX-5 engine—is priced at \$3,165. For greater horsepower requirements, the Parks is offered with an Anderson 150 h. p. engine at \$5,000 and with a Wright J-6, 165 h. p. at \$6,350. Our illustrated folder containing complete details will gladly be sent upon request.

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And you glide gently to earth for a happy landing!

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The Irvin Air Chute is available in seat, lap or back pack types. All Irvin are identical in construction and are made in two grades of fine silk, one priced at \$350, the other at \$270. Every one, regardless of price, complies with the standard U. S. Government parachute drawings.

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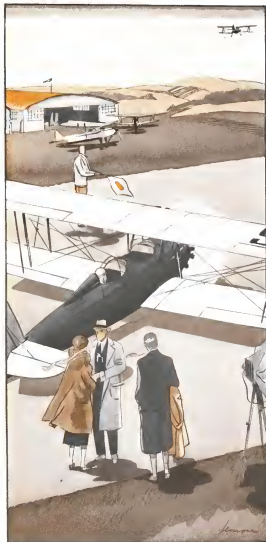








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